

Rocket Trajectory Nomograms

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Model rockets, A-C

Model rockets D-E

Mid Power, F-G

Small High Power, H-I

2inch High Power, I-J

3inch High Power, I-J

4inch High Power, J-K

6inch High Power, K-L

7.5inch High Power

Large rockets

A-C

D-E

F-G

H-I

2", I-J

3", I-J

4", J-K

6", K-L

7.5"

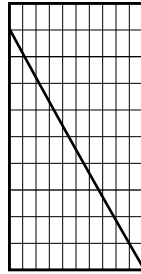
X

Kosdon-by-Aerotech

G135R

I_{tot} = 143.7 Ns
 F_{avg} = 137.0 N
 t_{burn} = 1.05 s
 d = 29 mm

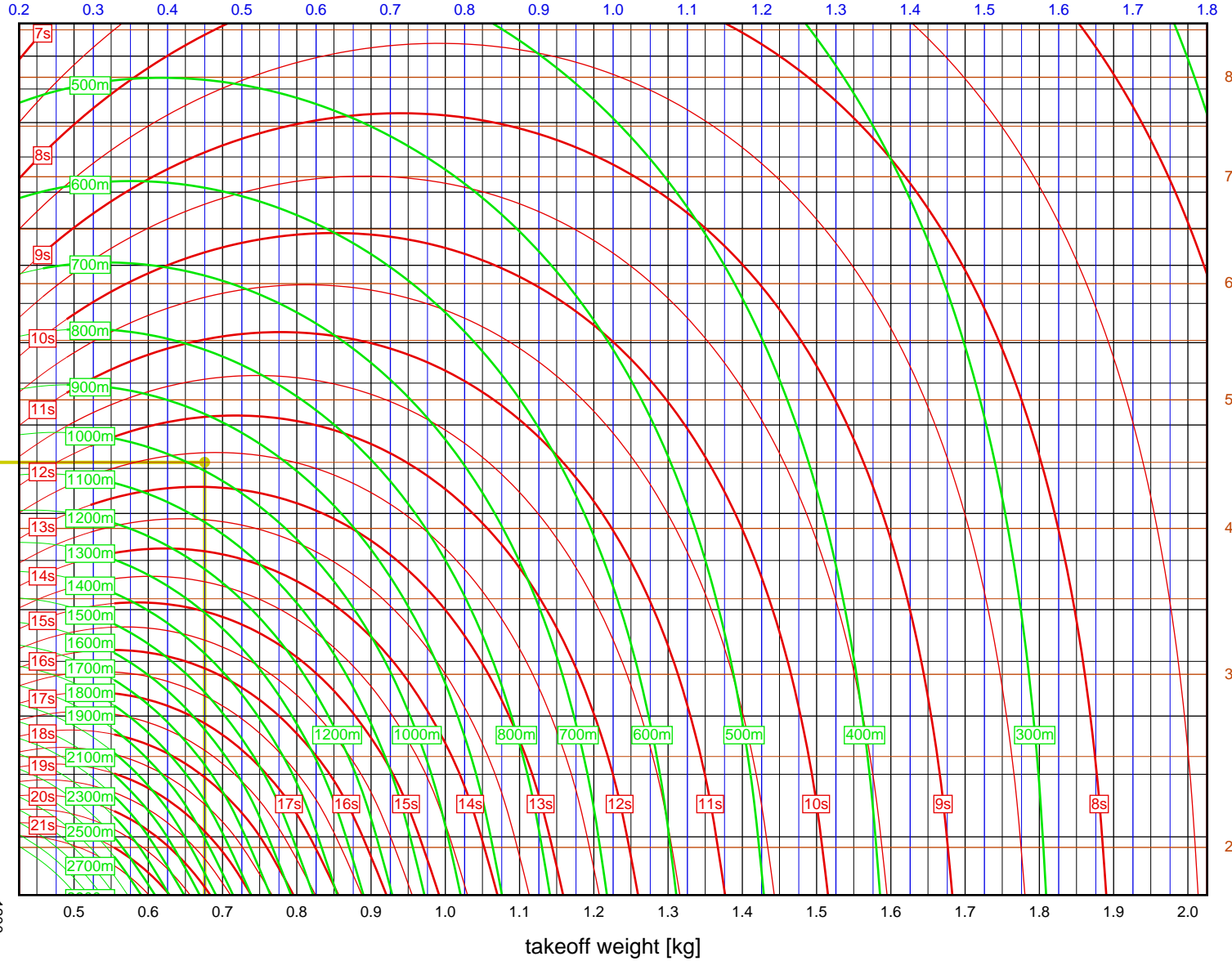
Data source:
Aerotech



1. From rocket diameter scale move down along slanted line to vertical line matching drag coefficient.
2. Move along horizontal to left border of density scale
3. Move up slanted line to vertical line matching density at launch site
4. From intersection point move horizontally to vertical line matching rocket mass
5. Read off expected time to apogee from red curves, altitude from green curves

Sample: diameter = 45mm, drag = 0.65, density = 1180 g/m³, weight = 0.676kg
 Results: time to apogee: 12.6s, expected altitude: 985m

empty weight [kg]



F-G 3

G135R

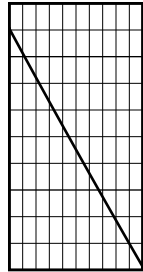
G135R

Kosdon-by-Aerotech

G82W

I_{tot} = 143.9 Ns
 F_{avg} = 73.4 N
 t_{burn} = 1.96 s
 d = 29 mm

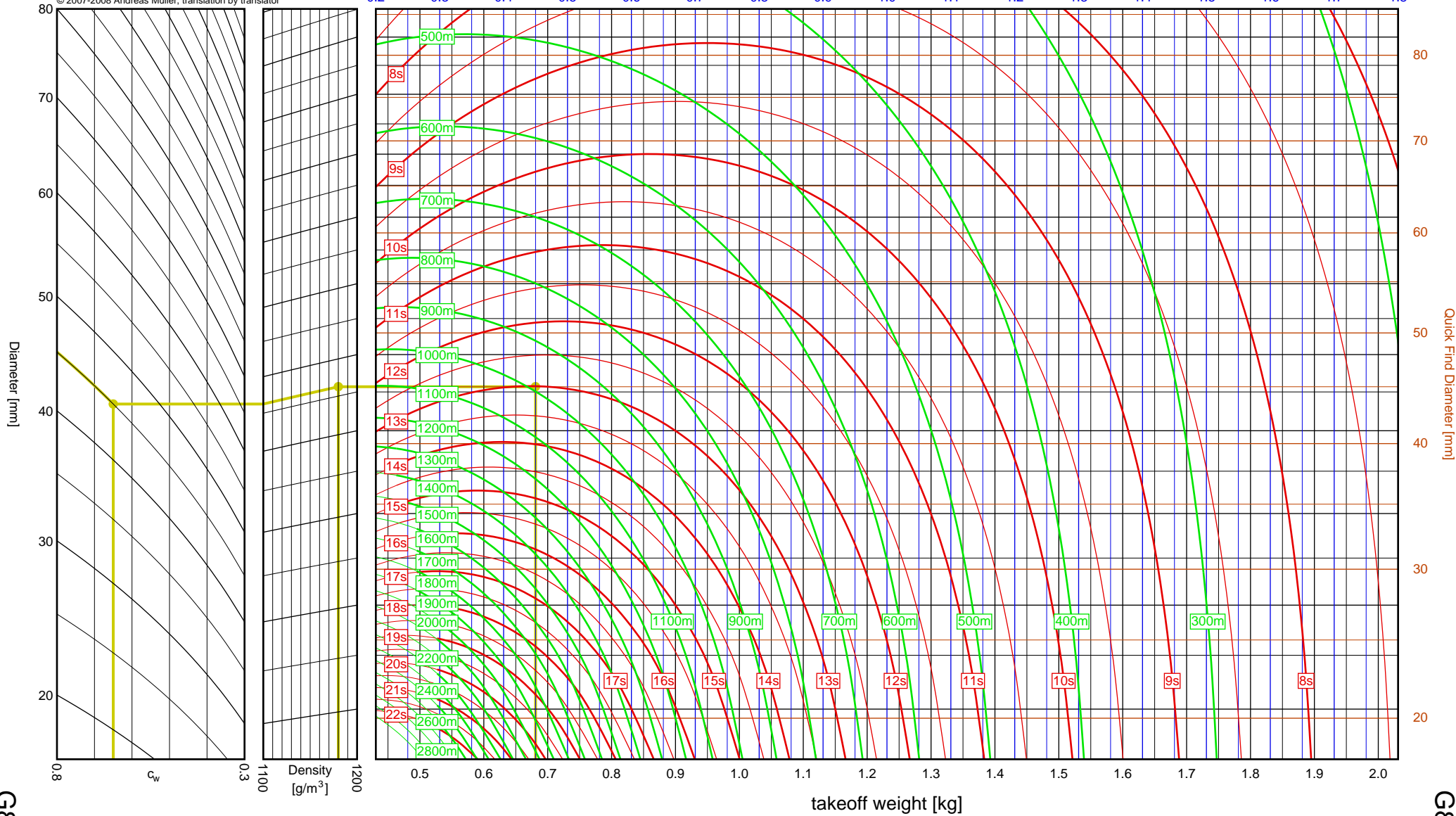
Data source:
Aerotech



1. From rocket diameter scale move down along slanted line to vertical line matching drag coefficient.
2. Move along horizontal to left border of density scale
3. Move up slanted line to vertical line matching density at launch site
4. From intersection point move horizontally to vertical line matching rocket mass
5. Read off expected time to apogee from red curves, altitude from green curves

Sample: diameter = 45mm, drag = 0.65, density = 1180 g/m³, weight = 0.681kg
Results: time to apogee: 13.0s, expected altitude: 991m

empty weight [kg]



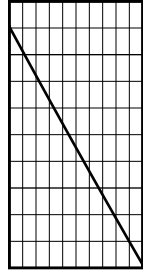
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Kosdon-by-Aerotech

H225R

I_{tot} = 242.7 Ns
 F_{avg} = 216.5 N
 t_{burn} = 1.12 s
 d = 29 mm

Data source:
Aerotech

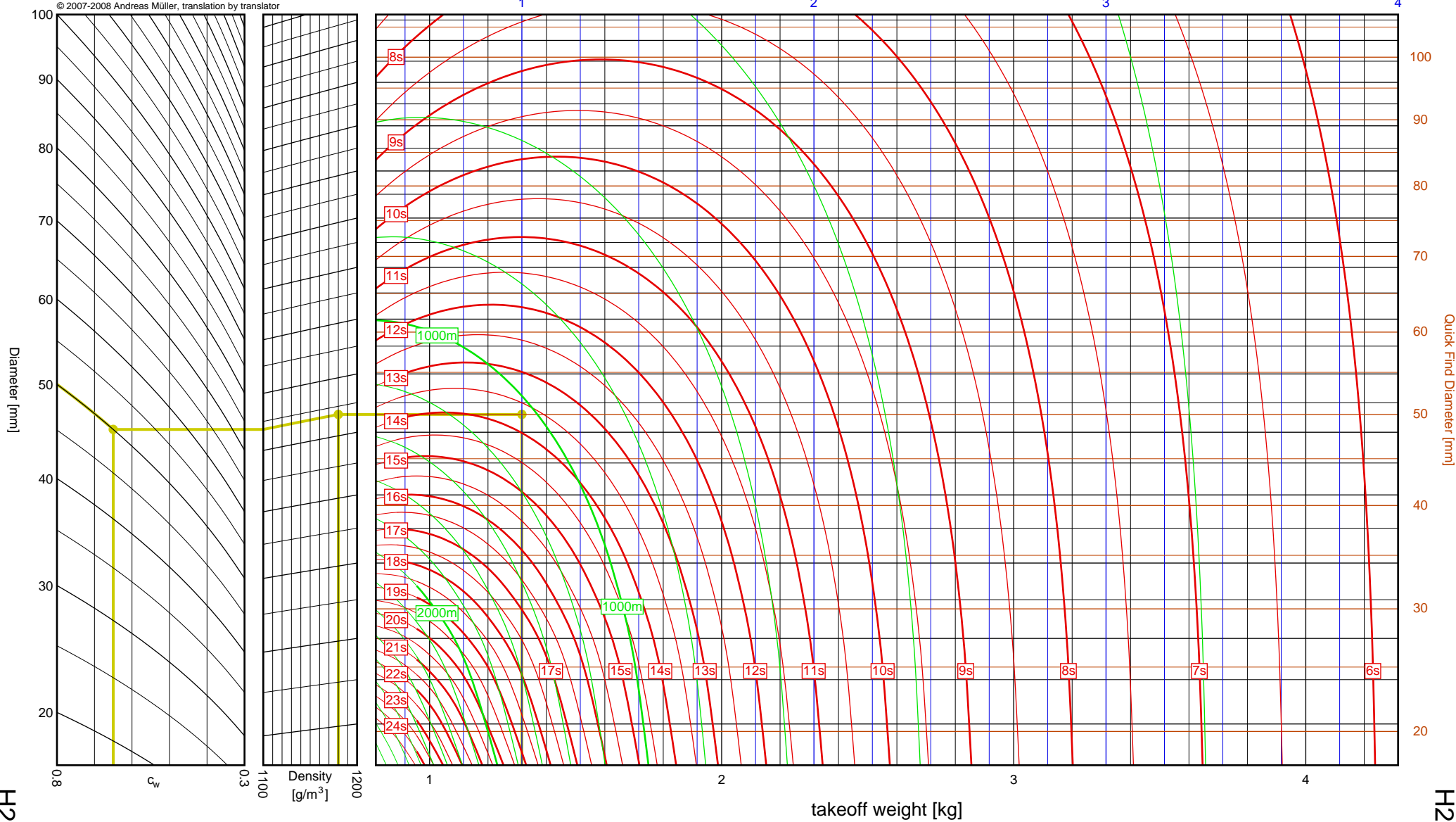


Launch site altitude [m ASL]
1000
500

1. From rocket diameter scale move down along slanted line to vertical line matching drag coefficient.
2. Move along horizontal to left border of density scale
3. Move up slanted line to vertical line matching density at launch site
4. From intersection point move horizontally to vertical line matching rocket mass
5. Read off expected time to apogee from red curves, altitude from green curves

Sample: diameter = 50mm, drag = 0.65, density = 1180 g/m³, weight = 1.316kg
Results: time to apogee: 13.7s, expected altitude: 1037m

empty weight [kg]



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H225R

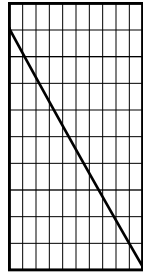
H225R

Kosdon-by-Aerotech

H130W

I_{tot} = 247.8 Ns
 F_{avg} = 129.8 N
 t_{burn} = 1.91 s
 d = 29 mm

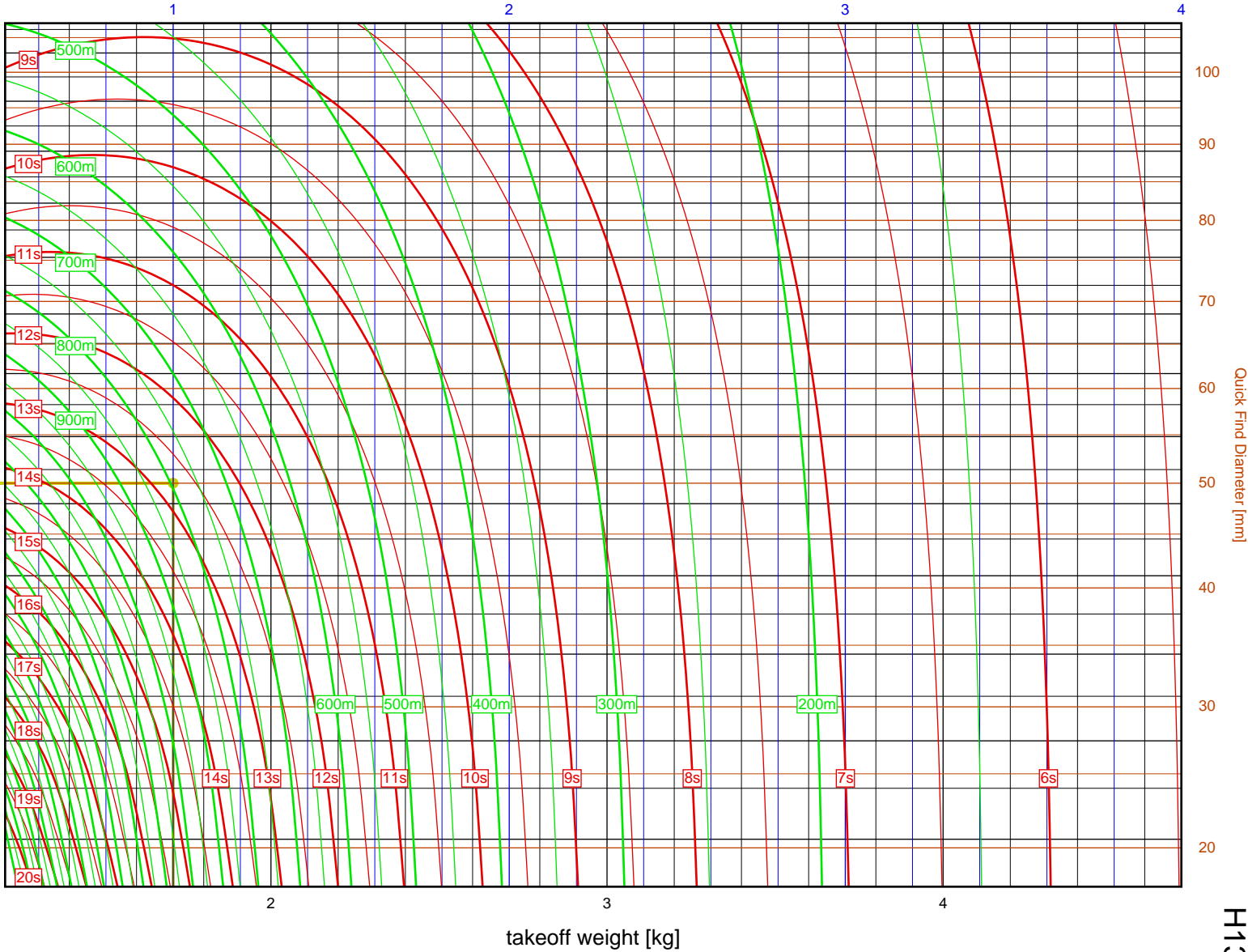
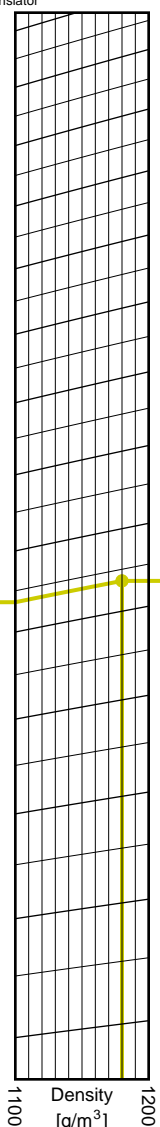
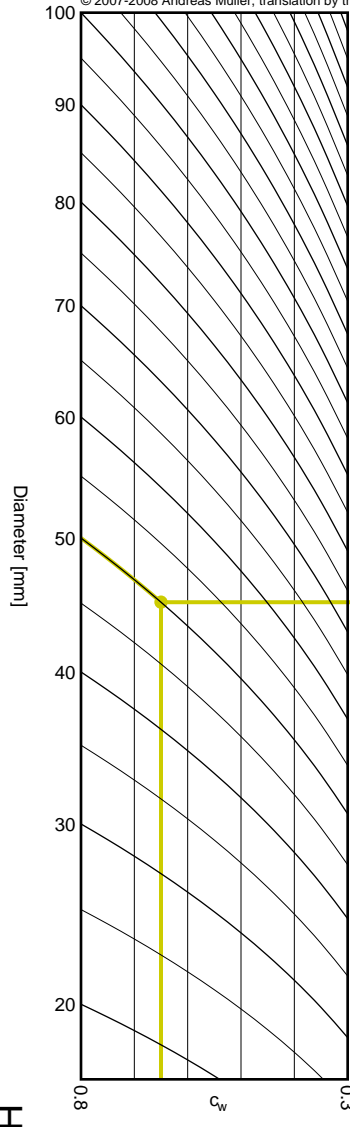
Data source:
Aerotech



1. From rocket diameter scale move down along slanted line to vertical line matching drag coefficient.
2. Move along horizontal to left border of density scale
3. Move up slanted line to vertical line matching density at launch site
4. From intersection point move horizontally to vertical line matching rocket mass
5. Read off expected time to apogee from red curves, altitude from green curves

Sample: diameter = 50mm, drag = 0.65, density = 1180 g/m³, weight = 1.709kg
Results: time to apogee: 12.8s, expected altitude: 795m

empty weight [kg]



H130W

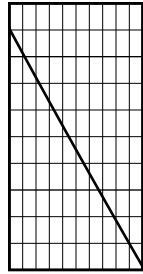
H130W

Kosdon-by-Aerotech

I170S

I_{tot} = 374.0 Ns
 F_{avg} = 175.6 N
 t_{burn} = 2.13 s
 d = 38 mm

Data source:
 Aerotech



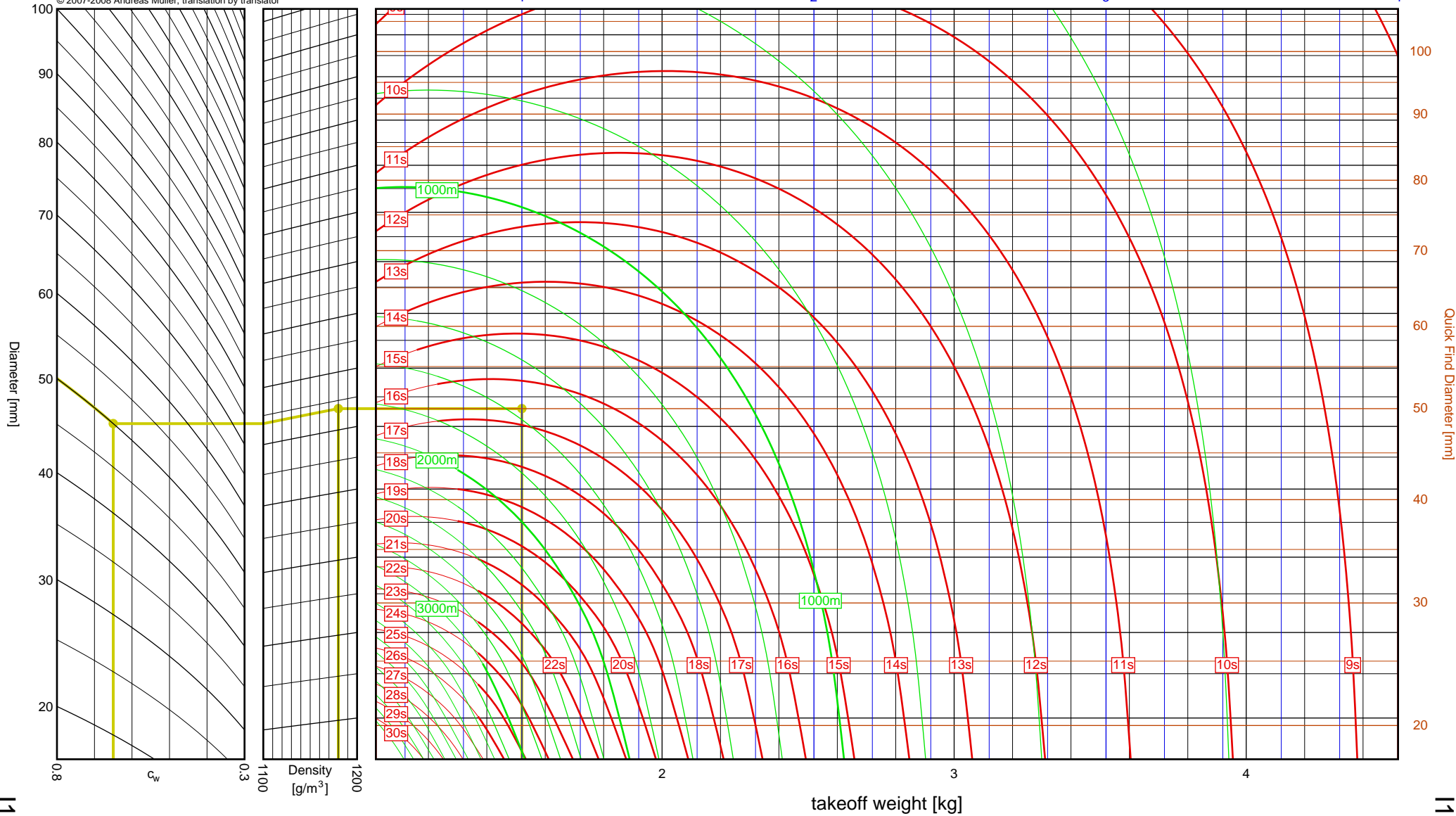
Launch site altitude [m ASL]

1. From rocket diameter scale move down along slanted line to vertical line matching drag coefficient.
2. Move along horizontal to left border of density scale
3. Move up slanted line to vertical line matching density at launch site
4. From intersection point move horizontally to vertical line matching rocket mass
5. Read off expected time to apogee from red curves, altitude from green curves

Sample: diameter = 50mm, drag = 0.65, density = 1180 g/m³, weight = 1.520kg
 Results: time to apogee: 16.6s, expected altitude: 1562m

empty weight [kg]

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I170S

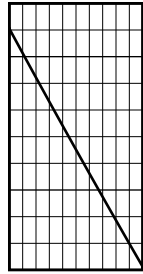
I170S

Kosdon-by-Aerotech

I301W

I_{tot} = 590.0 Ns
 F_{avg} = 302.2 N
 t_{burn} = 1.95 s
 d = 38 mm

Data source:
Aerotech

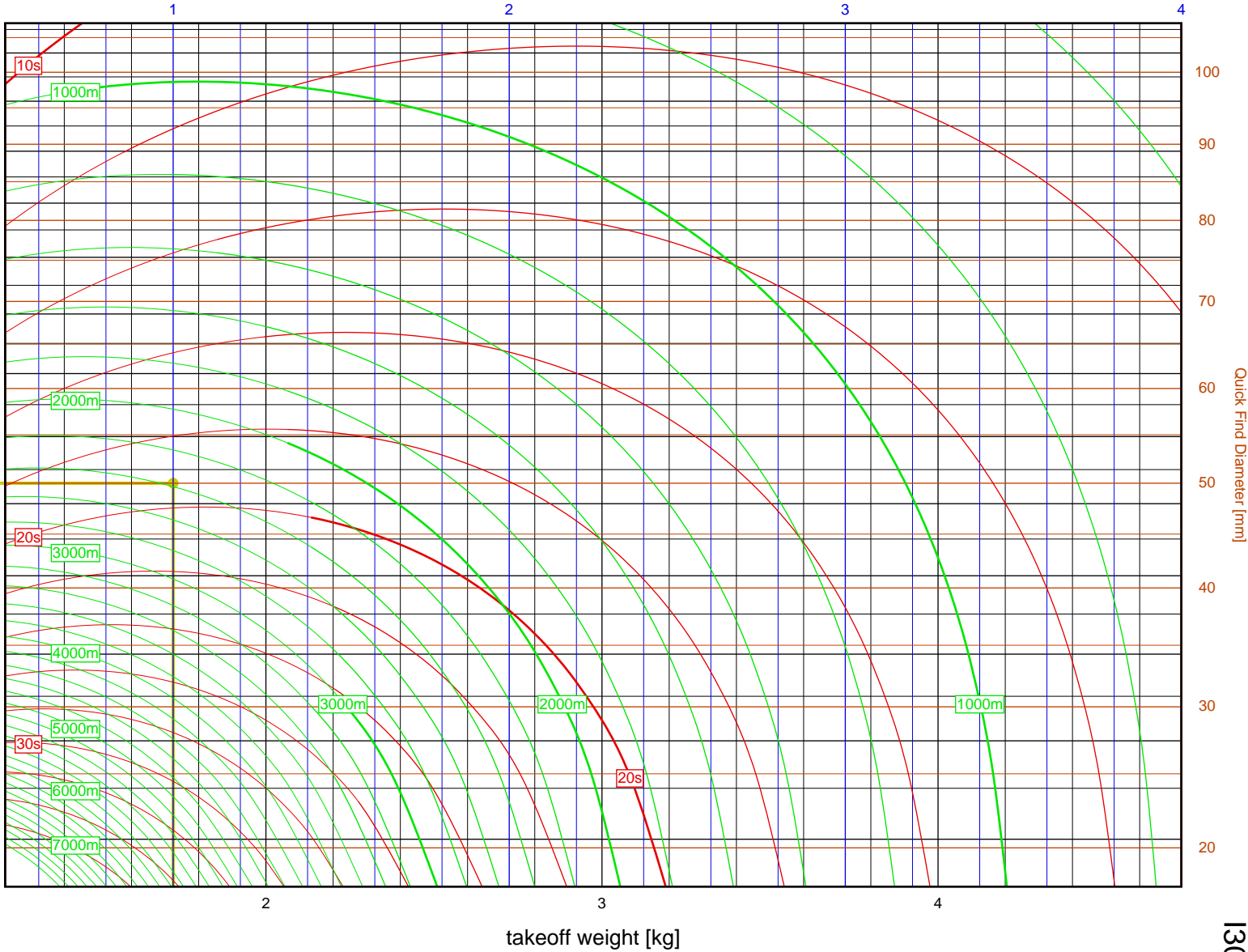
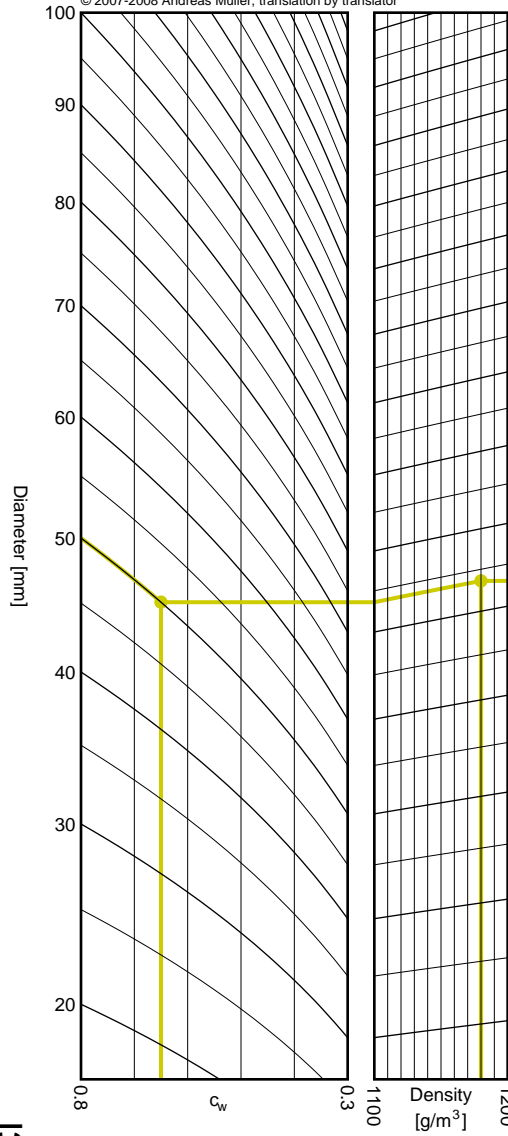


Launch site altitude [m ASL]

1. From rocket diameter scale move down along slanted line to vertical line matching drag coefficient.
2. Move along horizontal to left border of density scale
3. Move up slanted line to vertical line matching density at launch site
4. From intersection point move horizontally to vertical line matching rocket mass
5. Read off expected time to apogee from red curves, altitude from green curves

Sample: diameter = 50mm, drag = 0.65, density = 1180 g/m³, weight = 1.724kg
Results: time to apogee: 19.3s, expected altitude: 2383m

empty weight [kg]



I301W

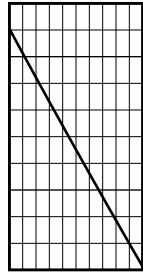
I301W

Kosdon-by-Aerotech

I550R

I_{tot} = 591.3 Ns
 F_{avg} = 543.0 N
 t_{burn} = 1.09 s
 d = 38 mm

Data source:
Aerotech

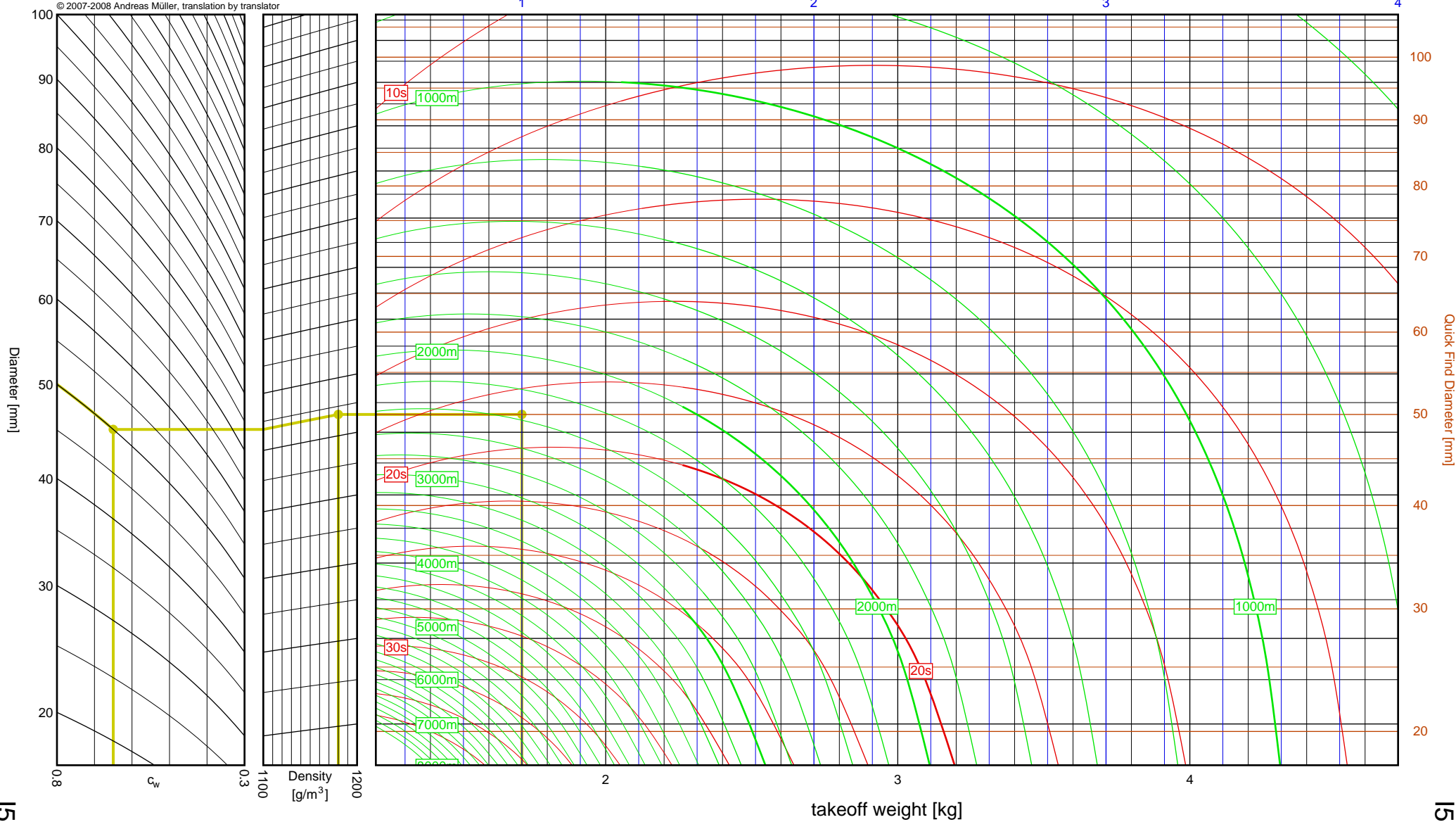


Launch site altitude [m ASL]

1. From rocket diameter scale move down along slanted line to vertical line matching drag coefficient.
2. Move along horizontal to left border of density scale
3. Move up slanted line to vertical line matching density at launch site
4. From intersection point move horizontally to vertical line matching rocket mass
5. Read off expected time to apogee from red curves, altitude from green curves

Sample: diameter = 50mm, drag = 0.65, density = 1180 g/m³, weight = 1.713kg
Results: time to apogee: 18.8s, expected altitude: 2363m

empty weight [kg]



I550R

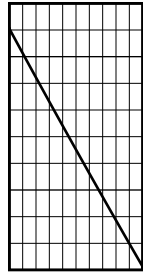
I550R

Kosdon-by-Aerotech

I170S

I_{tot} = 374.0 Ns
 F_{avg} = 175.6 N
 t_{burn} = 2.13 s
 d = 38 mm

Data source:
Aerotech

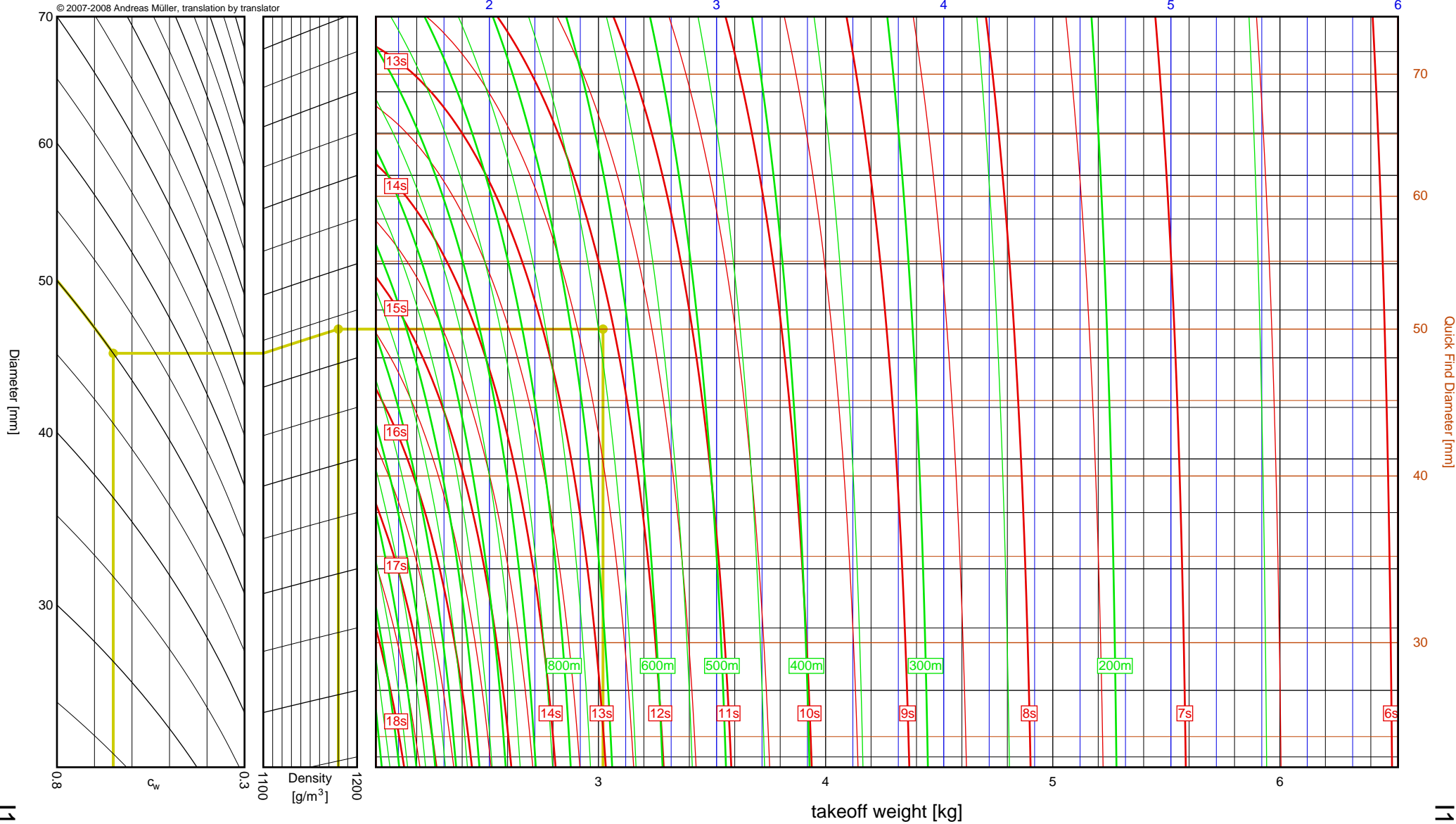


Launch site altitude [m ASL]

1. From rocket diameter scale move down along slanted line to vertical line matching drag coefficient.
2. Move along horizontal to left border of density scale
3. Move up slanted line to vertical line matching density at launch site
4. From intersection point move horizontally to vertical line matching rocket mass
5. Read off expected time to apogee from red curves, altitude from green curves

Sample: diameter = 50mm, drag = 0.65, density = 1180 g/m³, weight = 3.020kg
Results: time to apogee: 12.1s, expected altitude: 642m

empty weight [kg]



2", I-J

I170S

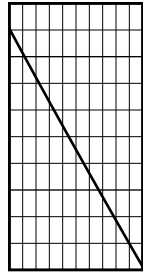
I170S

Kosdon-by-Aerotech

I301W

I_{tot} = 590.0 Ns
 F_{avg} = 302.2 N
 t_{burn} = 1.95 s
 d = 38 mm

Data source:
Aerotech

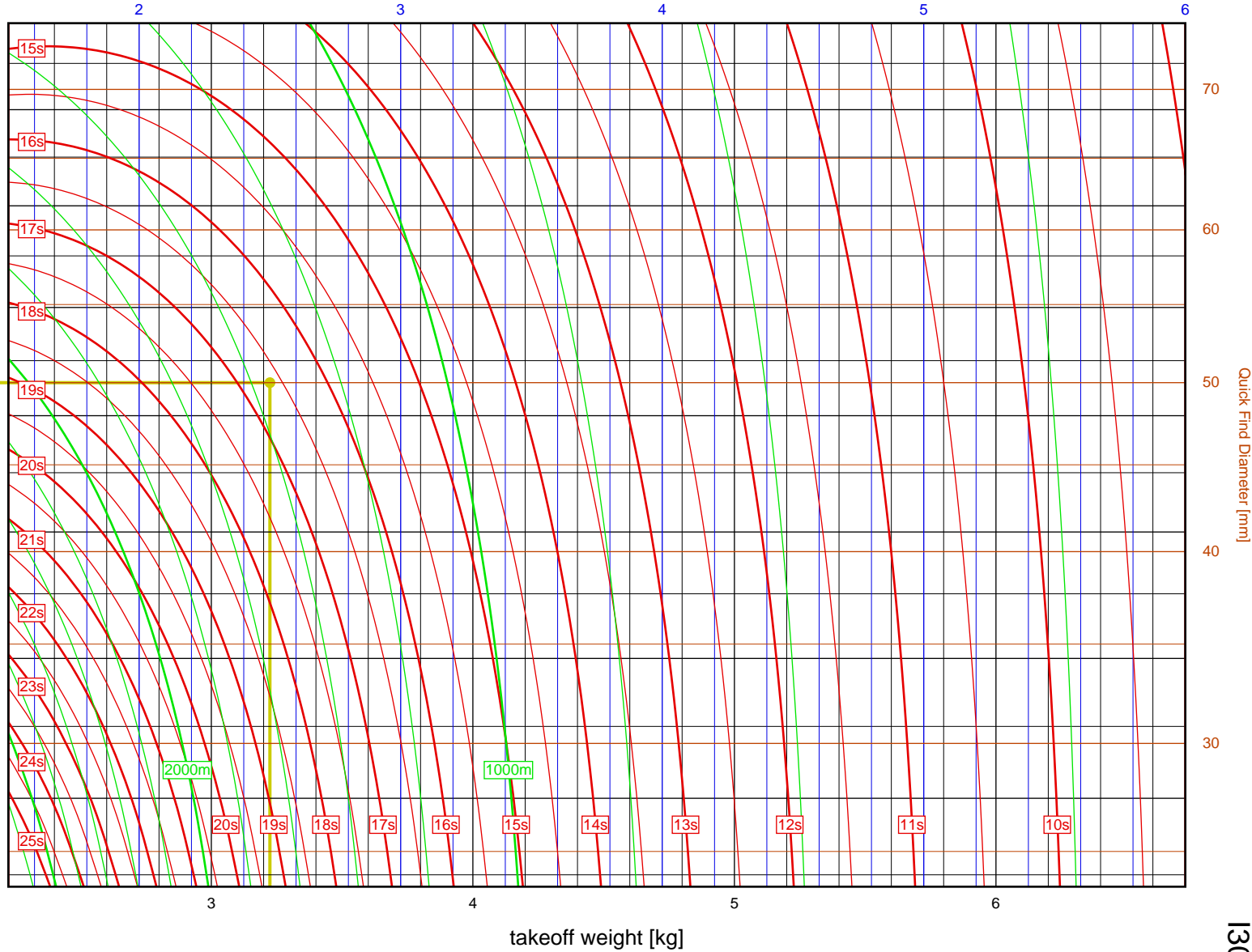
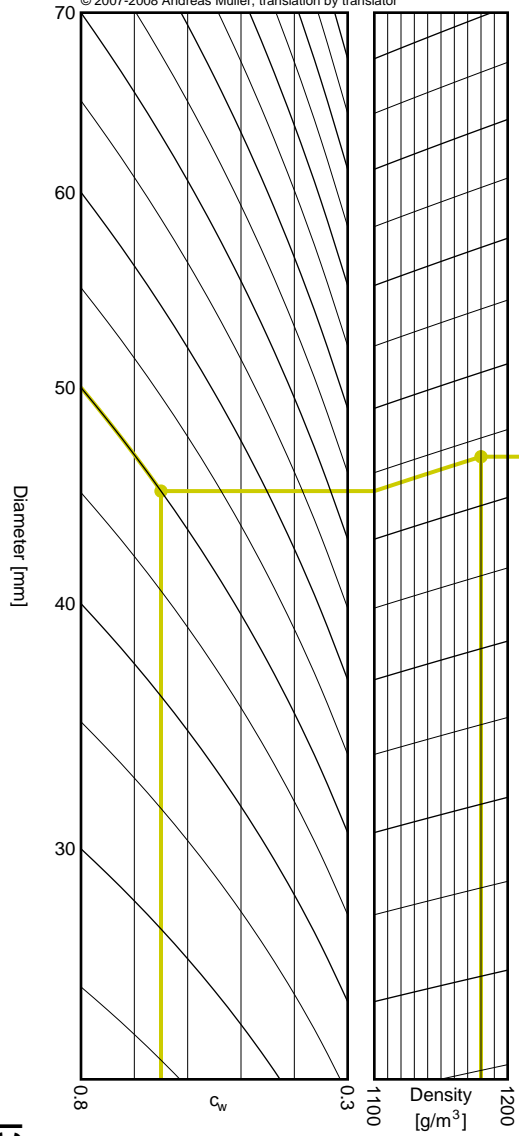


Launch site altitude [m ASL]

1. From rocket diameter scale move down along slanted line to vertical line matching drag coefficient.
2. Move along horizontal to left border of density scale
3. Move up slanted line to vertical line matching density at launch site
4. From intersection point move horizontally to vertical line matching rocket mass
5. Read off expected time to apogee from red curves, altitude from green curves

Sample: diameter = 50mm, drag = 0.65, density = 1180 g/m³, weight = 3.224kg
Results: time to apogee: 16.6s, expected altitude: 1356m

empty weight [kg]



I301W

I301W

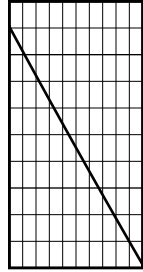
2", I-J

Kosdon-by-Aerotech

I550R

I_{tot} = 591.3 Ns
 F_{avg} = 543.0 N
 t_{burn} = 1.09 s
 d = 38 mm

Data source:
Aerotech

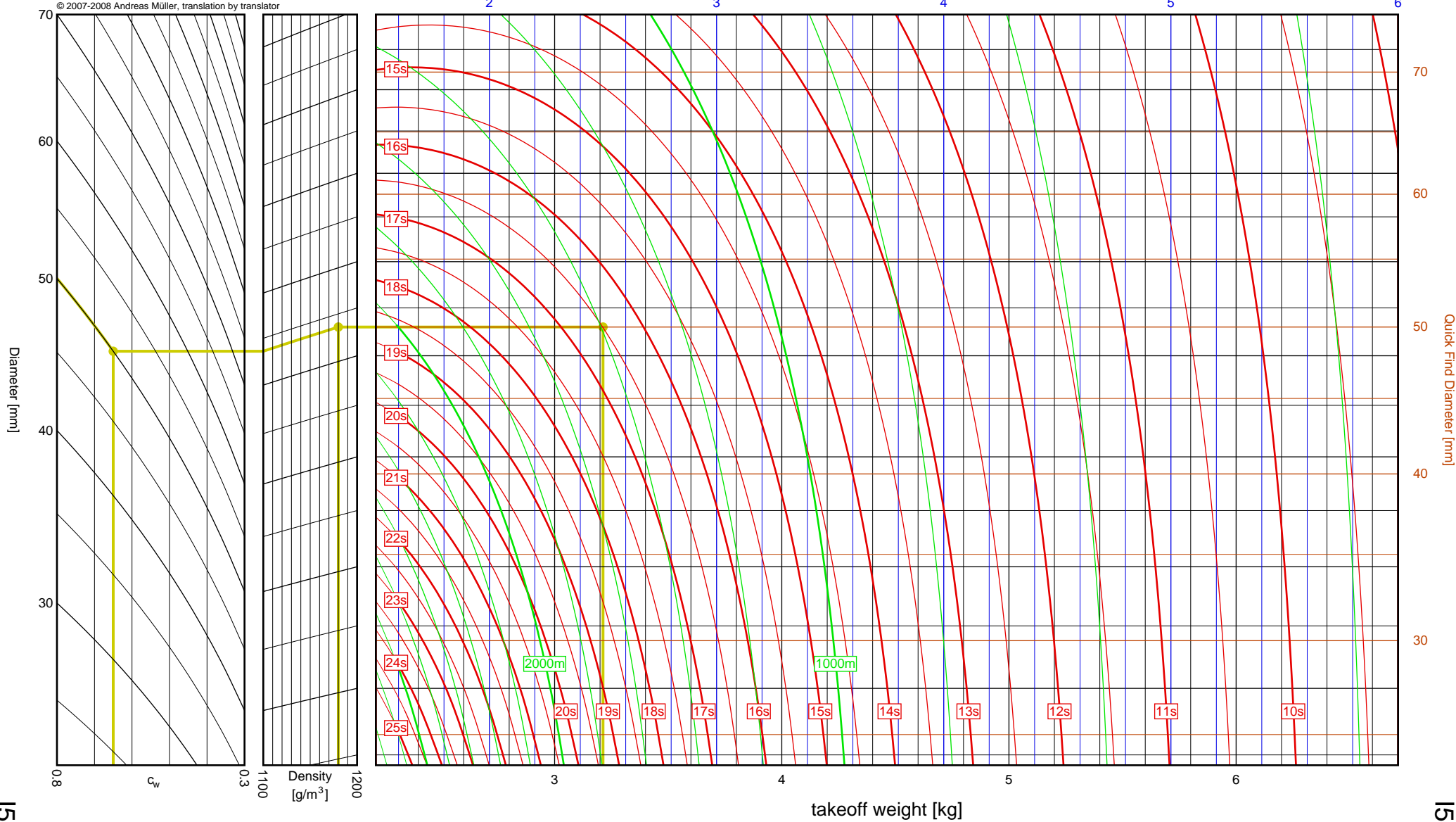


Launch site altitude [m ASL]

1. From rocket diameter scale move down along slanted line to vertical line matching drag coefficient.
2. Move along horizontal to left border of density scale
3. Move up slanted line to vertical line matching density at launch site
4. From intersection point move horizontally to vertical line matching rocket mass
5. Read off expected time to apogee from red curves, altitude from green curves

Sample: diameter = 50mm, drag = 0.65, density = 1180 g/m³, weight = 3.213kg
Results: time to apogee: 16.5s, expected altitude: 1395m

empty weight [kg]



I550R

I550R

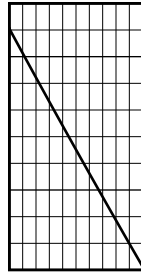
2", I-J

Kosdon-by-Aerotech

J740G

I_{tot} = 665.1 Ns
 F_{avg} = 788.1 N
 t_{burn} = 0.84 s
 d = 38 mm

Data source:
Aerotech

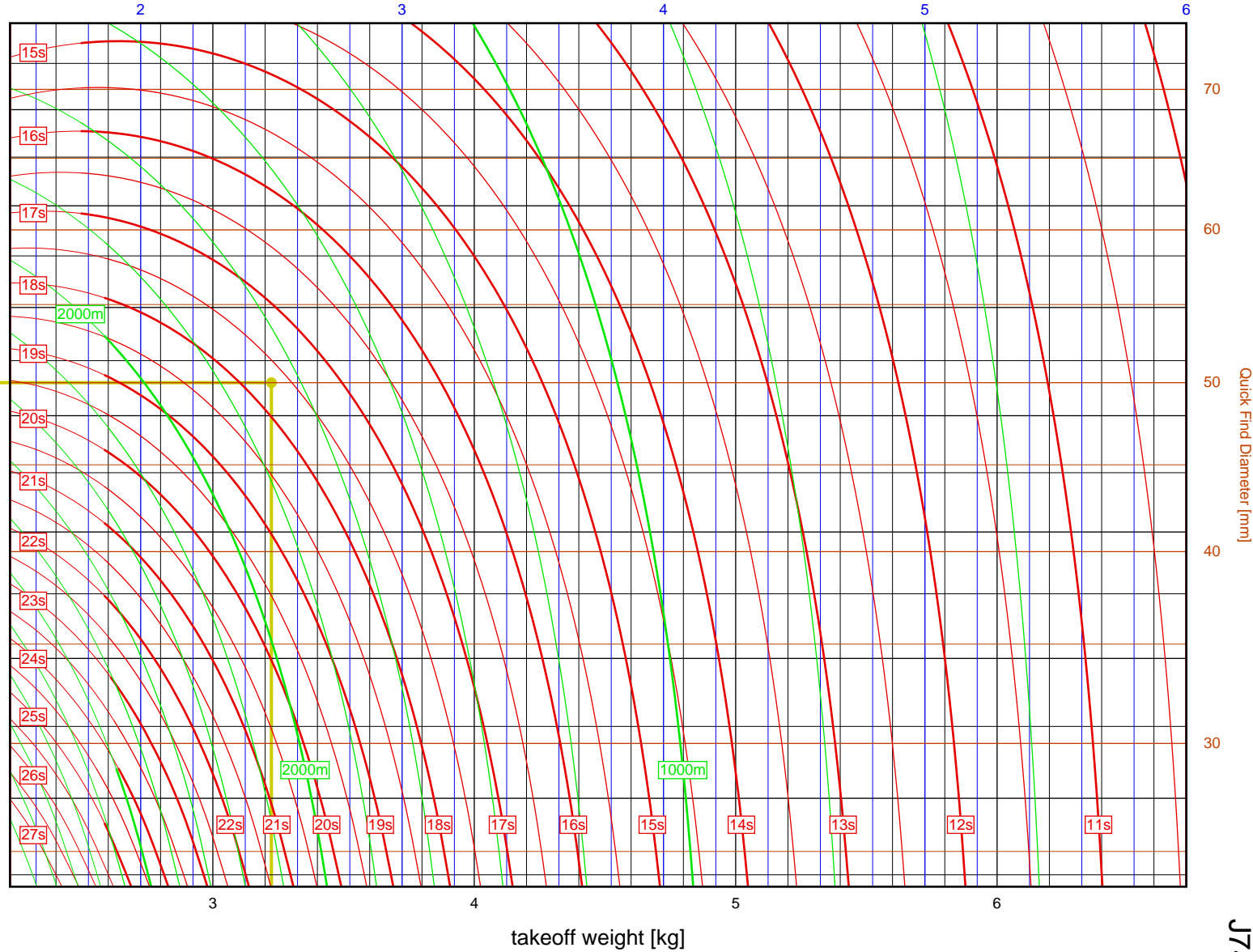
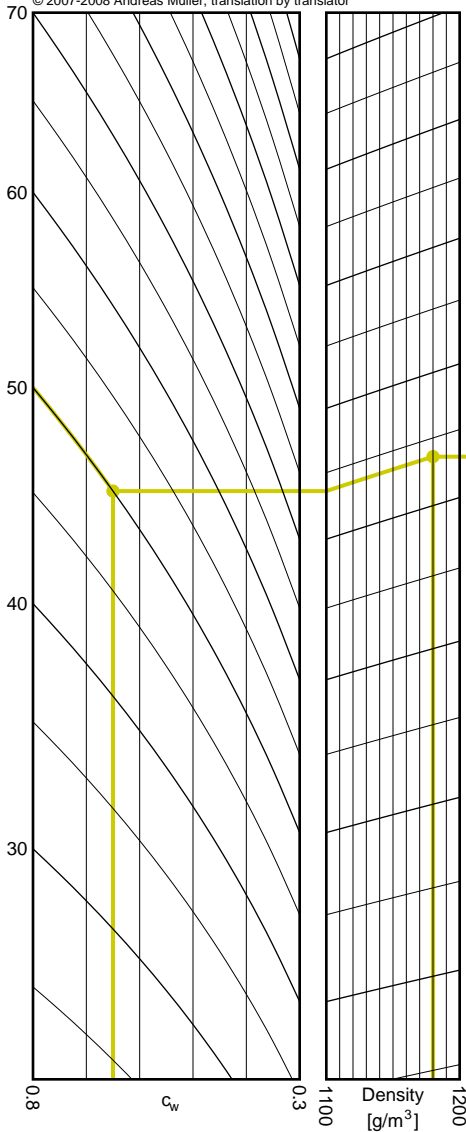


Launch site altitude [m ASL]

1. From rocket diameter scale move down along slanted line to vertical line matching drag coefficient.
2. Move along horizontal to left border of density scale
3. Move up slanted line to vertical line matching density at launch site
4. From intersection point move horizontally to vertical line matching rocket mass
5. Read off expected time to apogee from red curves, altitude from green curves

Sample: diameter = 50mm, drag = 0.65, density = 1180 g/m³, weight = 3.224kg
Results: time to apogee: 17.7s, expected altitude: 1672m

empty weight [kg]



2", I-J

J740G

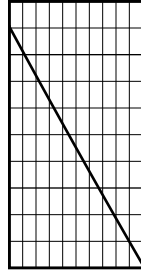
J740G

Kosdon-by-Aerotech

I170S

I_{tot} = 374.0 Ns
 F_{avg} = 175.6 N
 t_{burn} = 2.13 s
 d = 38 mm

Data source:
Aerotech

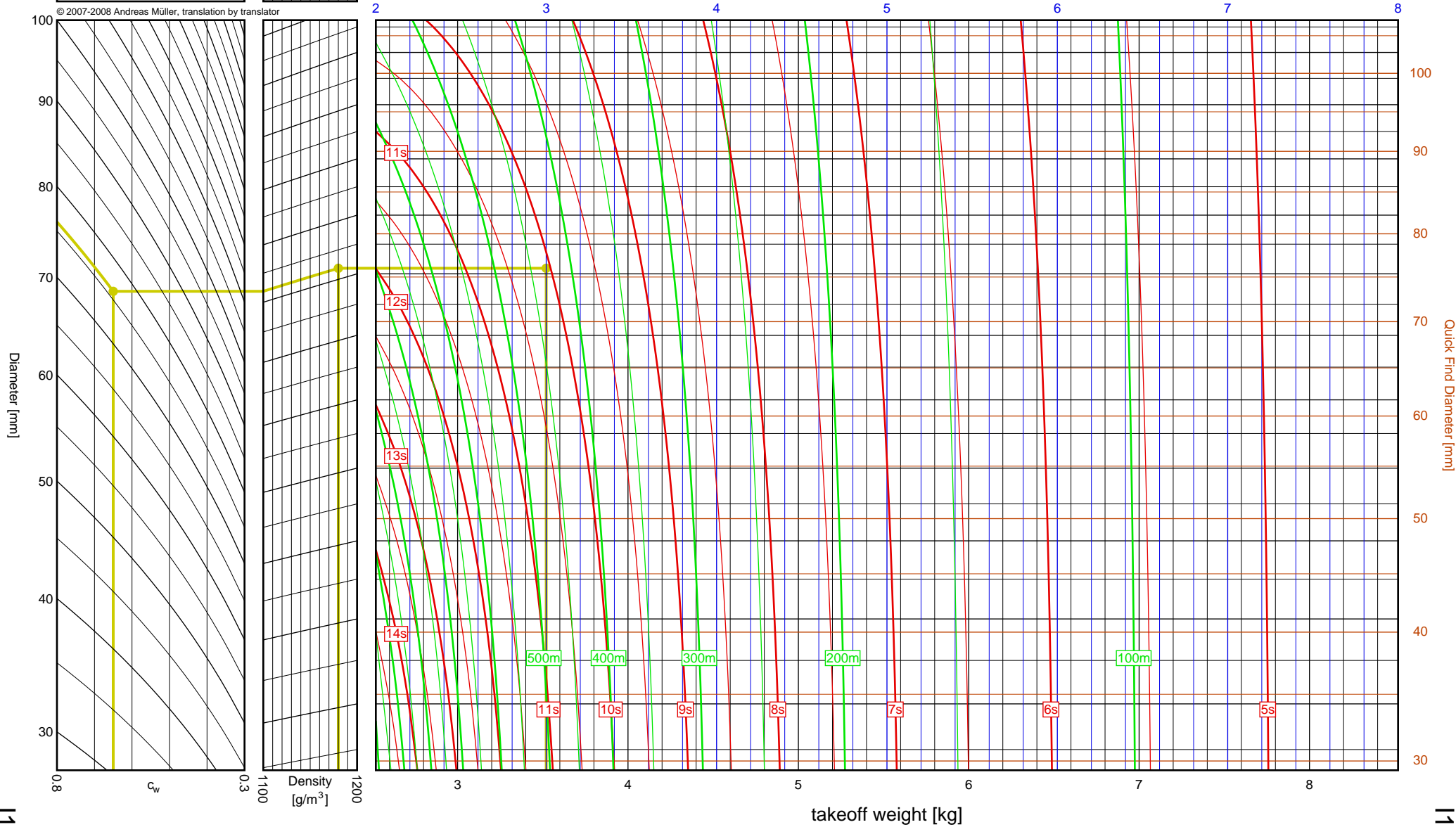


Launch site altitude [m ASL]

1. From rocket diameter scale move down along slanted line to vertical line matching drag coefficient.
2. Move along horizontal to left border of density scale
3. Move up slanted line to vertical line matching density at launch site
4. From intersection point move horizontally to vertical line matching rocket mass
5. Read off expected time to apogee from red curves, altitude from green curves

Sample: diameter = 76mm, drag = 0.65, density = 1180 g/m³, weight = 3.520kg
Results: time to apogee: 10.1s, expected altitude: 430m

empty weight [kg]



I170S

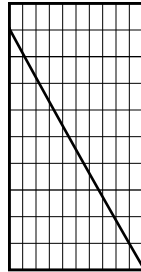
I170S

Kosdon-by-Aerotech

I301W

I_{tot} = 590.0 Ns
 F_{avg} = 302.2 N
 t_{burn} = 1.95 s
 d = 38 mm

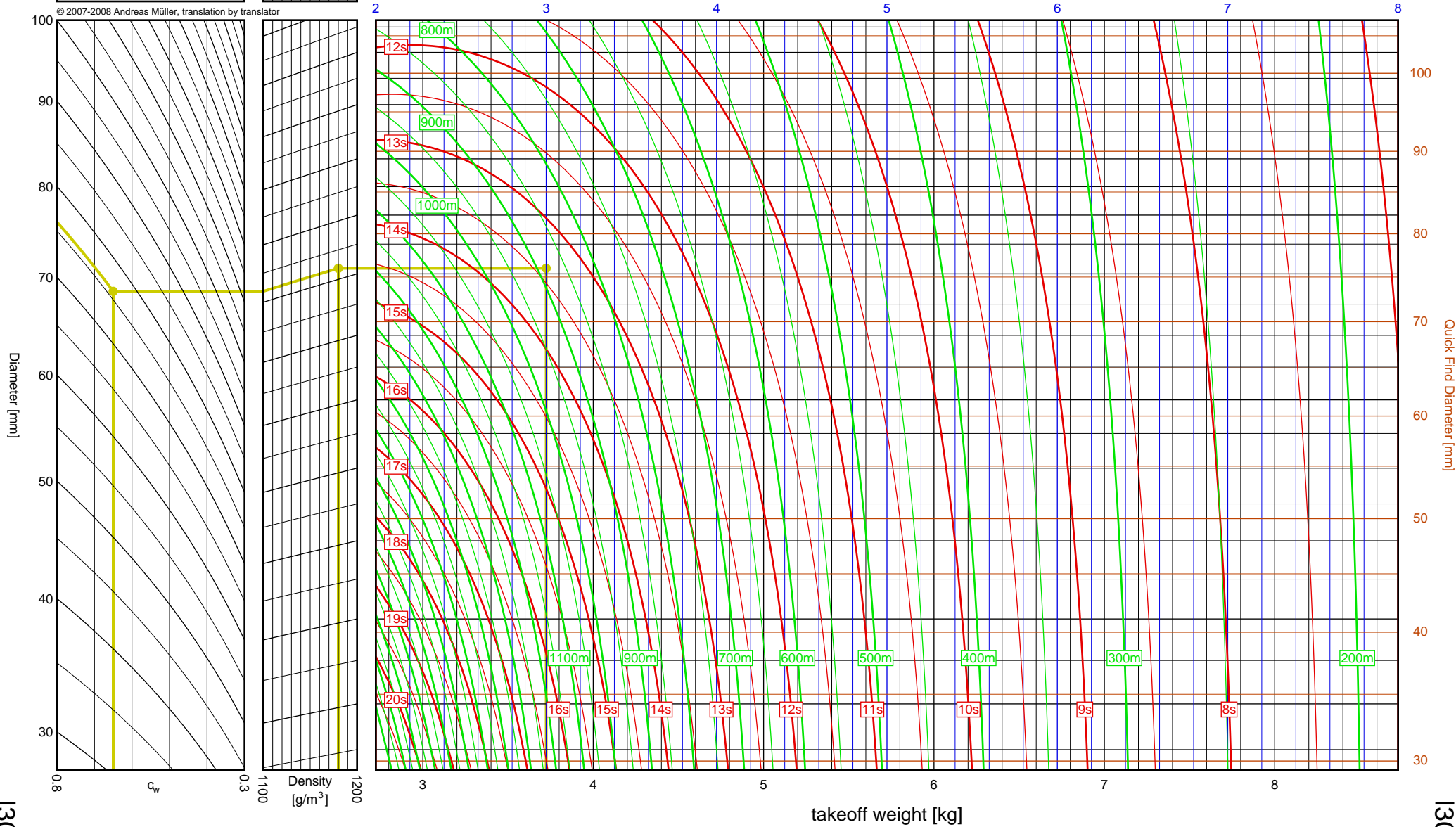
Data source:
Aerotech



1. From rocket diameter scale move down along slanted line to vertical line matching drag coefficient.
2. Move along horizontal to left border of density scale
3. Move up slanted line to vertical line matching density at launch site
4. From intersection point move horizontally to vertical line matching rocket mass
5. Read off expected time to apogee from red curves, altitude from green curves

Sample: diameter = 76mm, drag = 0.65, density = 1180 g/m³, weight = 3.724kg
Results: time to apogee: 13.4s, expected altitude: 884m

empty weight [kg]



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I301W

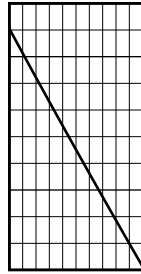
I301W

Kosdon-by-Aerotech

I550R

I_{tot} = 591.3 Ns
 F_{avg} = 543.0 N
 t_{burn} = 1.09 s
 d = 38 mm

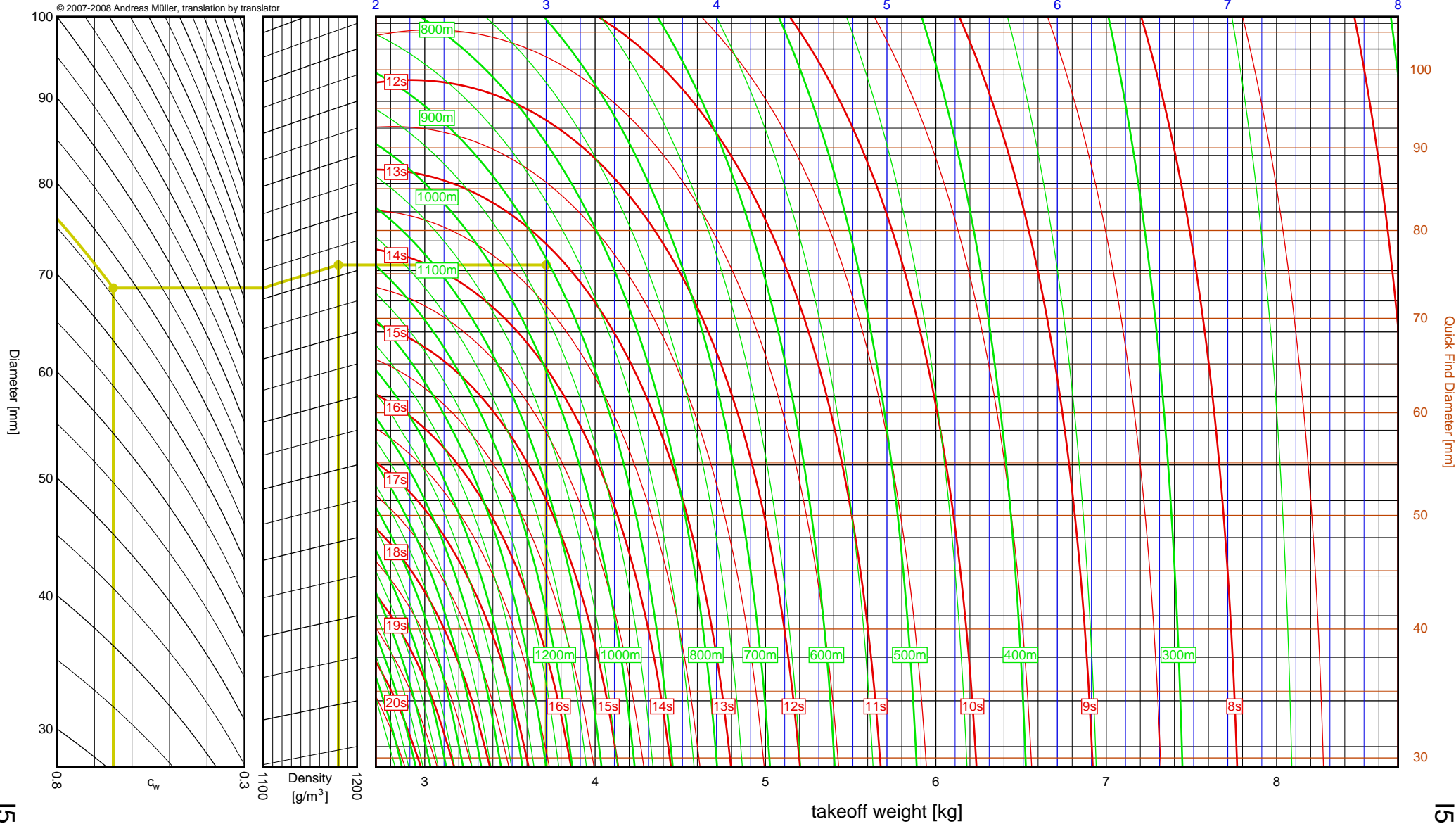
Data source:
Aerotech



1. From rocket diameter scale move down along slanted line to vertical line matching drag coefficient.
2. Move along horizontal to left border of density scale
3. Move up slanted line to vertical line matching density at launch site
4. From intersection point move horizontally to vertical line matching rocket mass
5. Read off expected time to apogee from red curves, altitude from green curves

Sample: diameter = 76mm, drag = 0.65, density = 1180 g/m³, weight = 3.713kg
Results: time to apogee: 13.2s, expected altitude: 906m

empty weight [kg]



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I550R

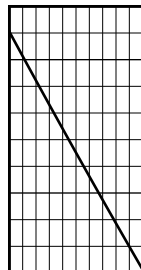
I550R

Kosdon-by-Aerotech

J740G

I_{tot} = 665.1 Ns
 F_{avg} = 788.1 N
 t_{burn} = 0.84 s
 d = 38 mm

Data source:
Aerotech

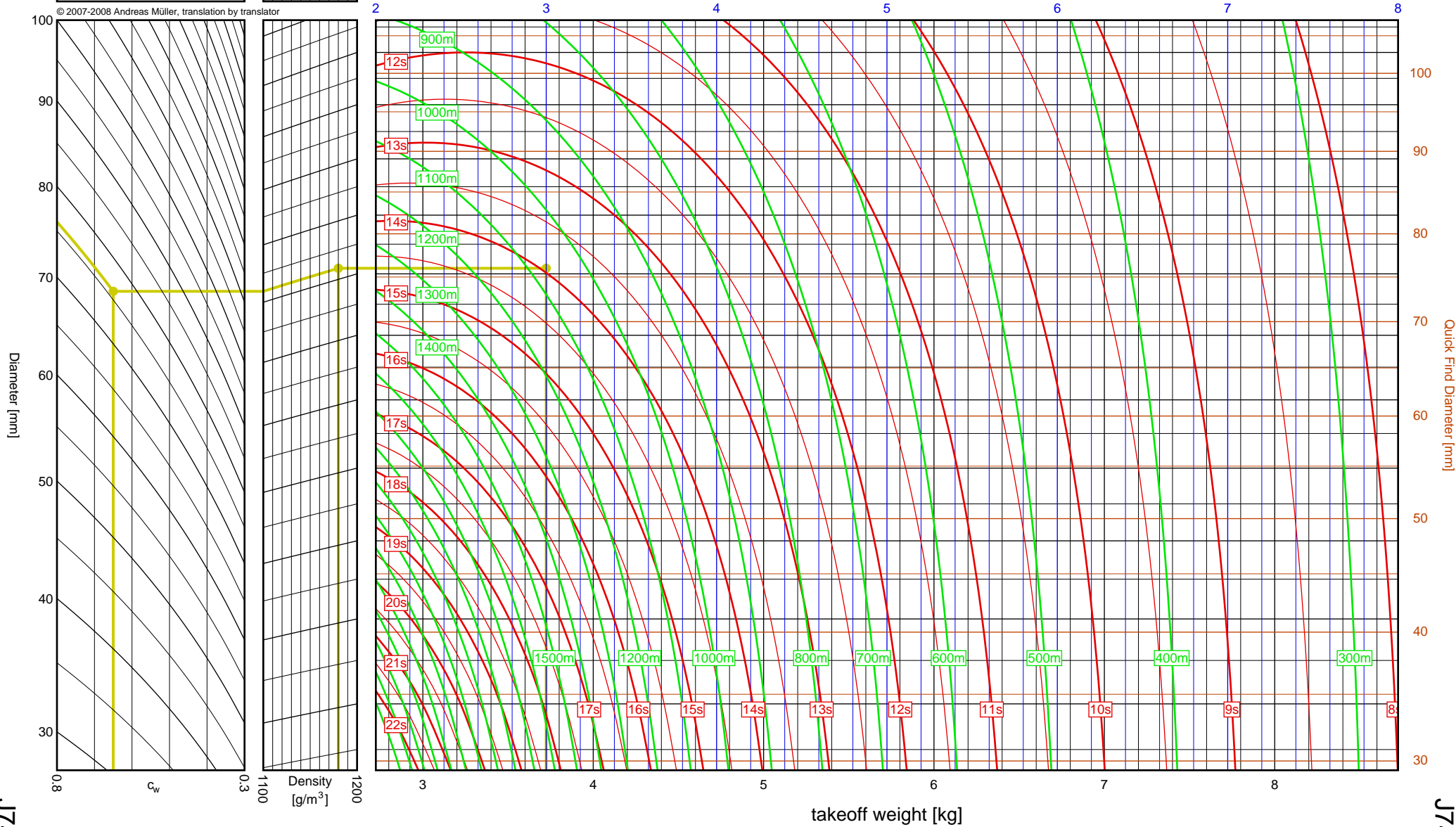


Launch site altitude (m ASL)

1. From rocket diameter scale move down along slanted line to vertical line matching drag coefficient.
2. Move along horizontal to left border of density scale
3. Move up slanted line to vertical line matching density at launch site
4. From intersection point move horizontally to vertical line matching rocket mass
5. Read off expected time to apogee from red curves, altitude from green curves

Sample: diameter = 76mm, drag = 0.65, density = 1180 g/m³, weight = 3.724kg
Results: time to apogee: 14.0s, expected altitude: 1069m

empty weight [kg]



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J740G

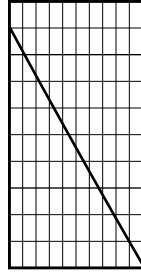
J740G

Kosdon-by-Aerotech K520F

I_{tot} = 645.5 Ns
 F_{avg} = 514.3 N
 t_{burn} = 1.25 s
 d = 38 mm

Data source:
Aerotech

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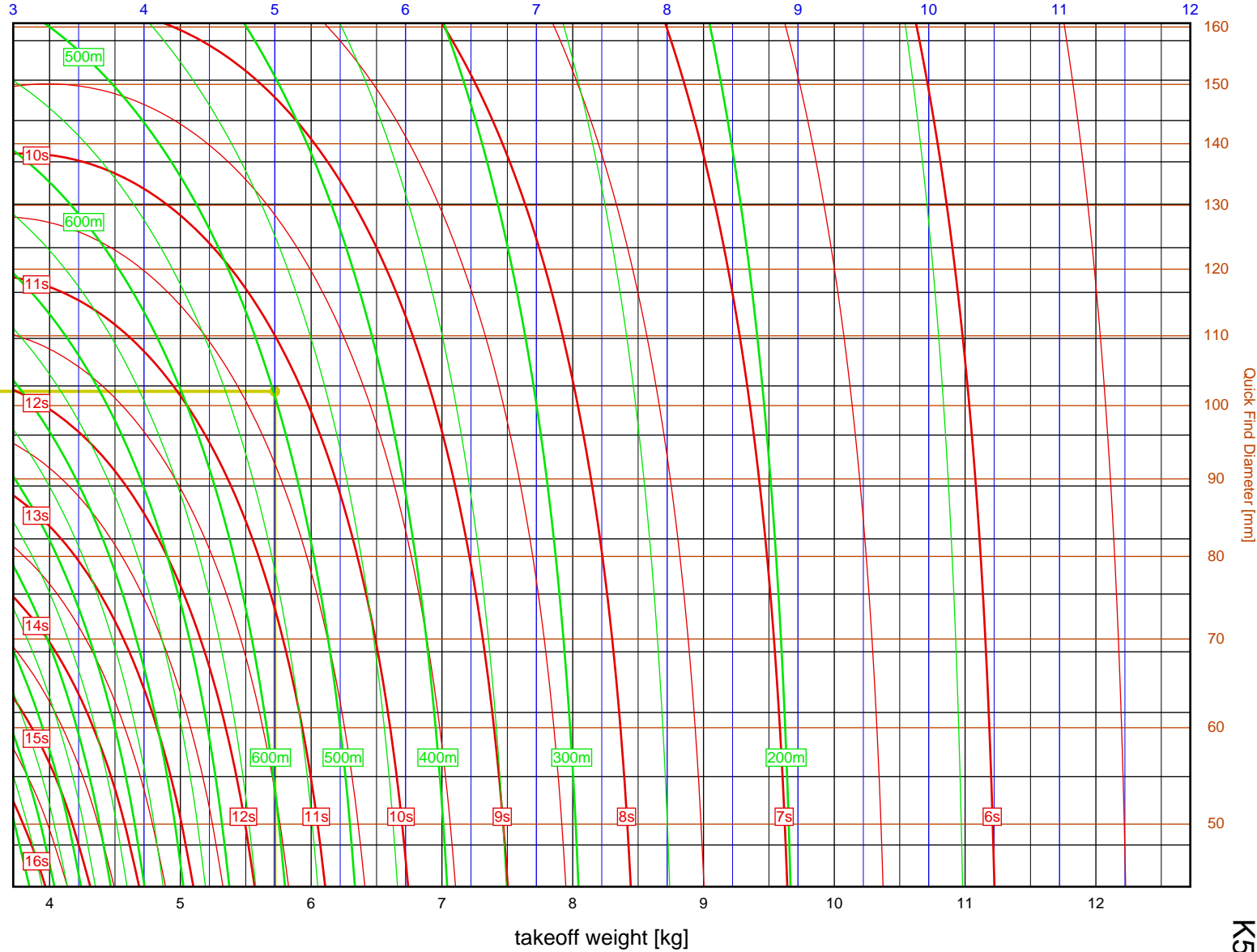
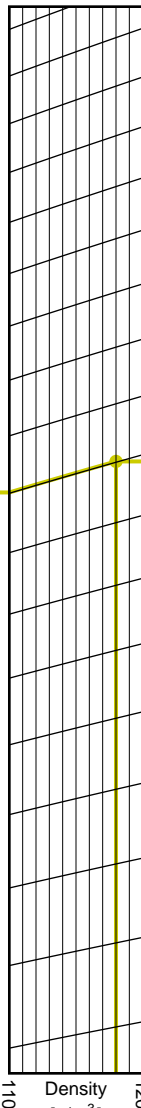
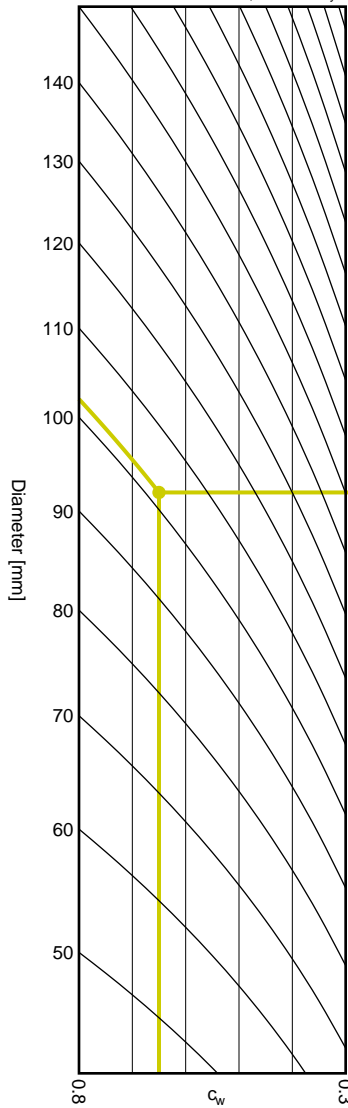


Launch site altitude [m ASL]

1. From rocket diameter scale move down along slanted line to vertical line matching drag coefficient.
2. Move along horizontal to left border of density scale
3. Move up slanted line to vertical line matching density at launch site
4. From intersection point move horizontally to vertical line matching rocket mass
5. Read off expected time to apogee from red curves, altitude from green curves

Sample: diameter = 102mm, drag = 0.65, density = 1180 g/m³, weight = 5.722kg
Results: time to apogee: 10.2s, expected altitude: 499m

empty weight [kg]



takeoff weight [kg]

4", J-K ⁷

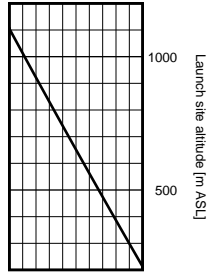
K520F

K520F

Kosdon-by-Aerotech J740G

I_{tot} = 665.1 Ns
 F_{avg} = 788.1 N
 t_{burn} = 0.84 s
 d = 38 mm

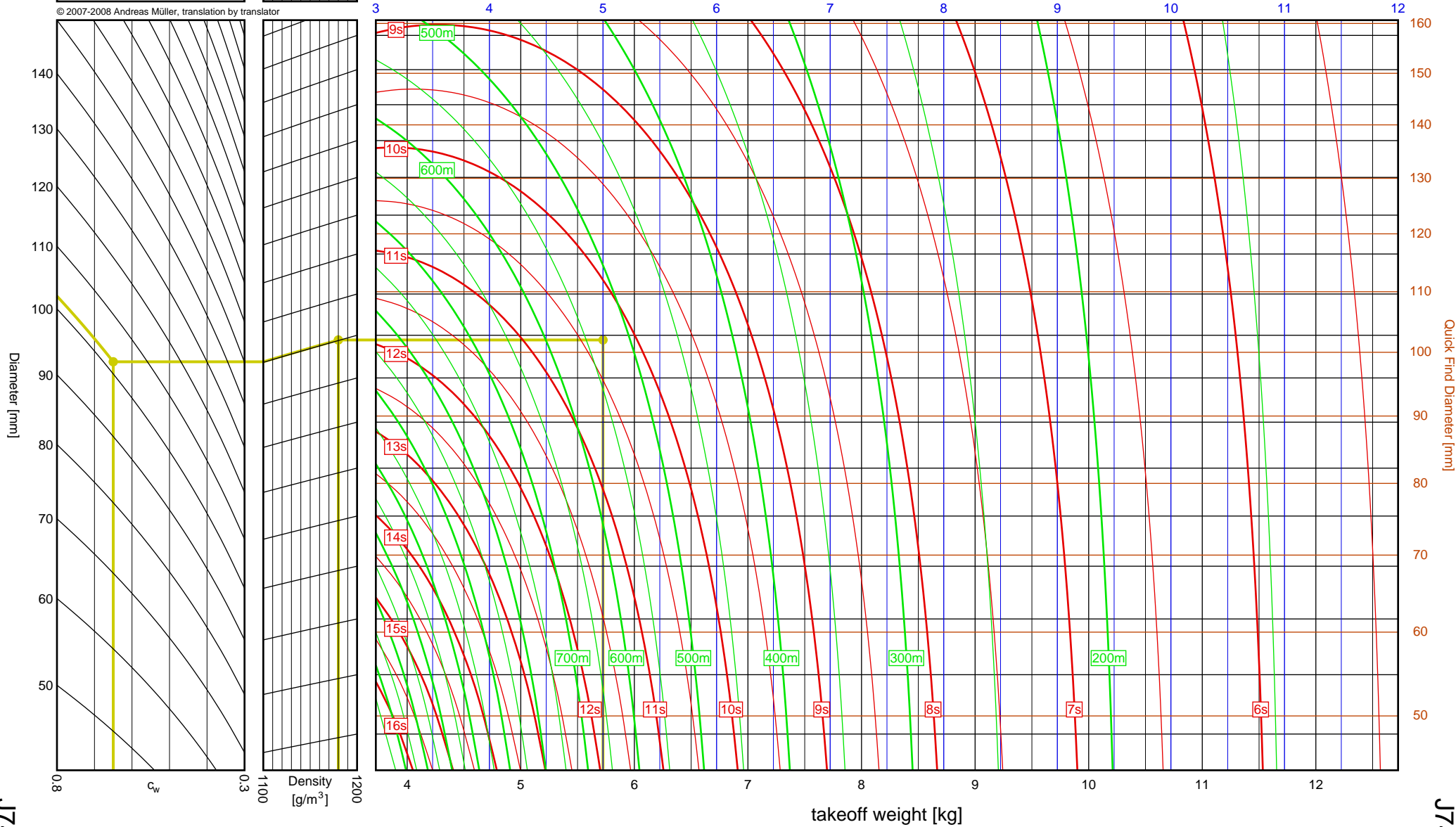
Data source:
Aerotech



1. From rocket diameter scale move down along slanted line to vertical line matching drag coefficient.
2. Move along horizontal to left border of density scale
3. Move up slanted line to vertical line matching density at launch site
4. From intersection point move horizontally to vertical line matching rocket mass
5. Read off expected time to apogee from red curves, altitude from green curves

Sample: diameter = 102mm, drag = 0.65, density = 1180 g/m³, weight = 5.724kg
Results: time to apogee: 10.3s, expected altitude: 531m

empty weight [kg]



takeoff weight [kg]

4", J-K ⁷

J740G

J740G

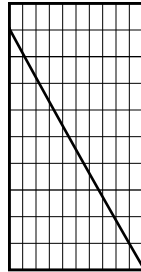
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Kosdon-by-Aerotech

K700F

I_{tot} = 1436.0 Ns
 F_{avg} = 776.2 N
 t_{burn} = 1.85 s
 d = 54 mm

Data source:
Aerotech

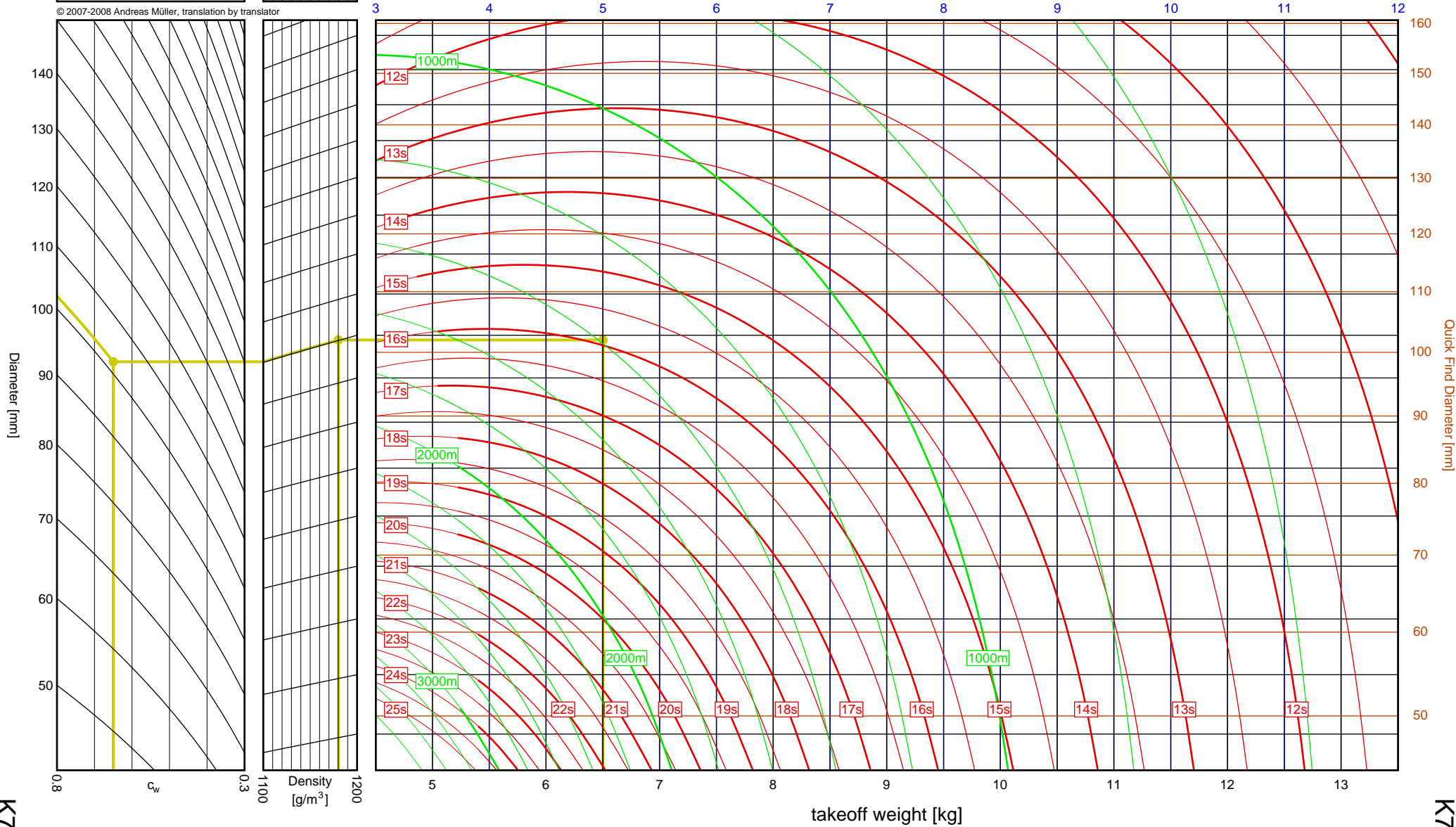


Launch site altitude [m ASL]

1. From rocket diameter scale move down along slanted line to vertical line matching drag coefficient.
2. Move along horizontal to left border of density scale
3. Move up slanted line to vertical line matching density at launch site
4. From intersection point move horizontally to vertical line matching rocket mass
5. Read off expected time to apogee from red curves, altitude from green curves

Sample: diameter = 102mm, drag = 0.65, density = 1180 g/m³, weight = 6.503kg
Results: time to apogee: 15.9s, expected altitude: 1394m

empty weight [kg]



takeoff weight [kg]

4", J-K ⁷

K700F

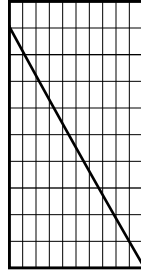
K700F

Kosdon-by-Aerotech

K700F

I_{tot} = 1436.0 Ns
 F_{avg} = 776.2 N
 t_{burn} = 1.85 s
 d = 54 mm

Data source:
Aerotech



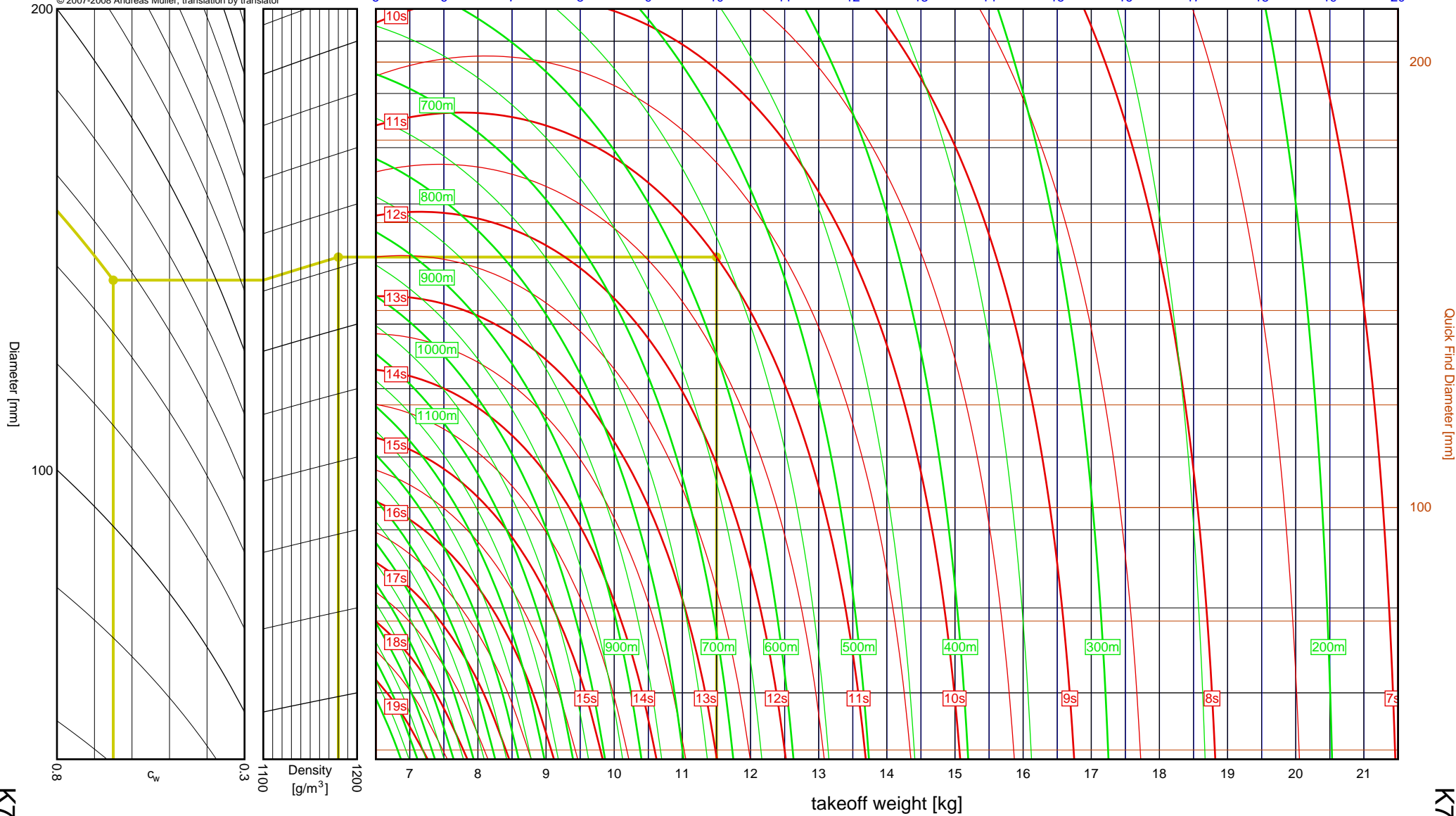
Launch site altitude [m ASL]

1. From rocket diameter scale move down along slanted line to vertical line matching drag coefficient.
2. Move along horizontal to left border of density scale
3. Move up slanted line to vertical line matching density at launch site
4. From intersection point move horizontally to vertical line matching rocket mass
5. Read off expected time to apogee from red curves, altitude from green curves

Sample: diameter = 152mm, drag = 0.65, density = 1180 g/m³, weight = 11.503kg
Results: time to apogee: 11.0s, expected altitude: 559m

empty weight [kg]

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6", K-L⁸

K700F

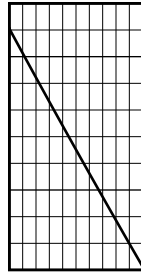
K700F

Kosdon-by-Aerotech

K1750R

$I_{tot} = 2468.8 \text{ Ns}$
 $F_{avg} = 1690.9 \text{ N}$
 $t_{burn} = 1.46 \text{ s}$
 $d = 54 \text{ mm}$

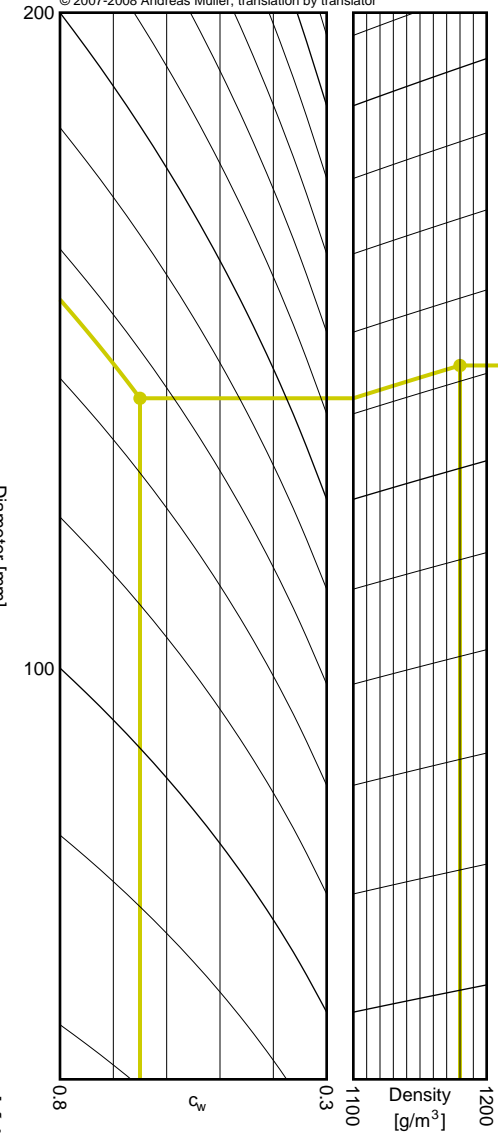
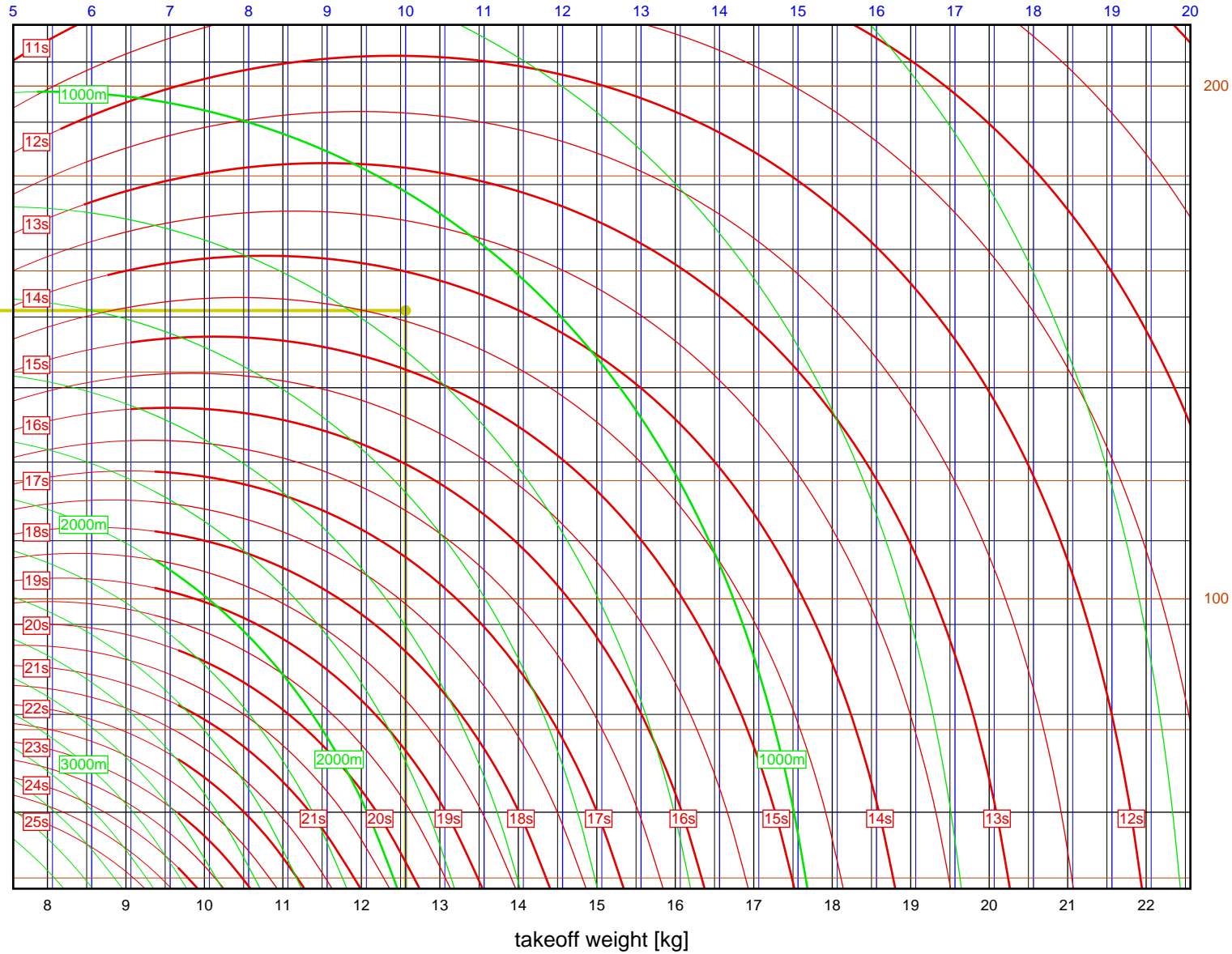
Data source:
Aerotech



1. From rocket diameter scale move down along slanted line to vertical line matching drag coefficient.
2. Move along horizontal to left border of density scale
3. Move up slanted line to vertical line matching density at launch site
4. From intersection point move horizontally to vertical line matching rocket mass
5. Read off expected time to apogee from red curves, altitude from green curves

Sample: diameter = 152mm, drag = 0.65, density = 1180 g/m³, weight = 12.564kg
Results: time to apogee: 14.4s, expected altitude: 1147m

empty weight [kg]



K1750R

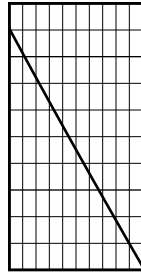
K1750R

Kosdon-by-Aerotech

K750W

I_{tot} = 2538.5 Ns
 F_{avg} = 732.0 N
 t_{burn} = 3.47 s
 d = 54 mm

Data source:
Aerotech

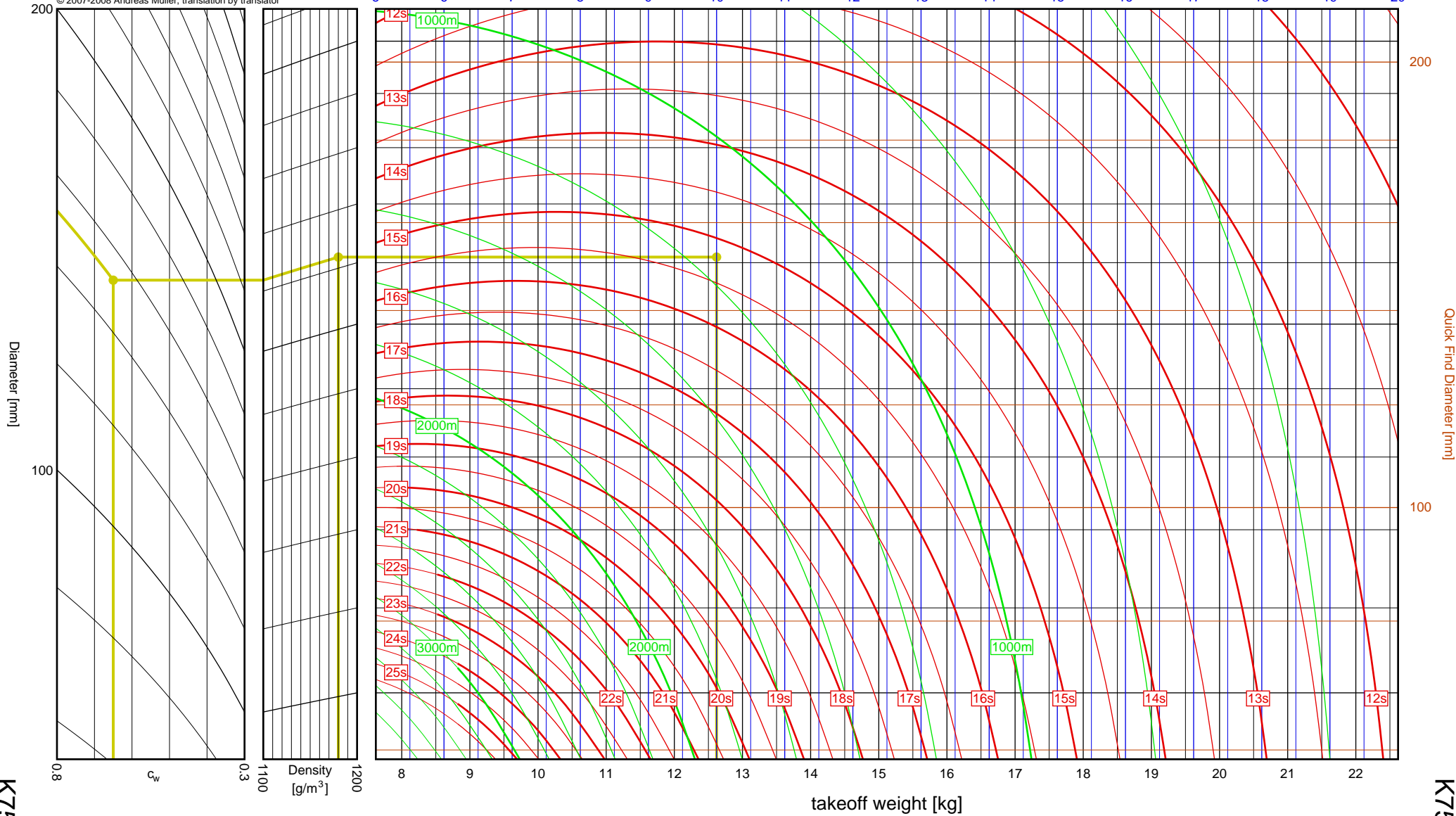


Launch site altitude [m ASL]

1. From rocket diameter scale move down along slanted line to vertical line matching drag coefficient.
2. Move along horizontal to left border of density scale
3. Move up slanted line to vertical line matching density at launch site
4. From intersection point move horizontally to vertical line matching rocket mass
5. Read off expected time to apogee from red curves, altitude from green curves

Sample: diameter = 152mm, drag = 0.65, density = 1180 g/m³, weight = 12.620kg
Results: time to apogee: 15.2s, expected altitude: 1160m

empty weight [kg]



K750W

K750W

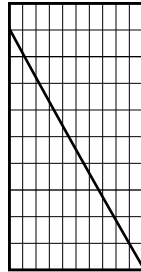
6", K-L⁸

Kosdon-by-Aerotech

L2300G

I_{tot} = 2737.0 Ns
 F_{avg} = 2191.4 N
 t_{burn} = 1.25 s
 d = 54 mm

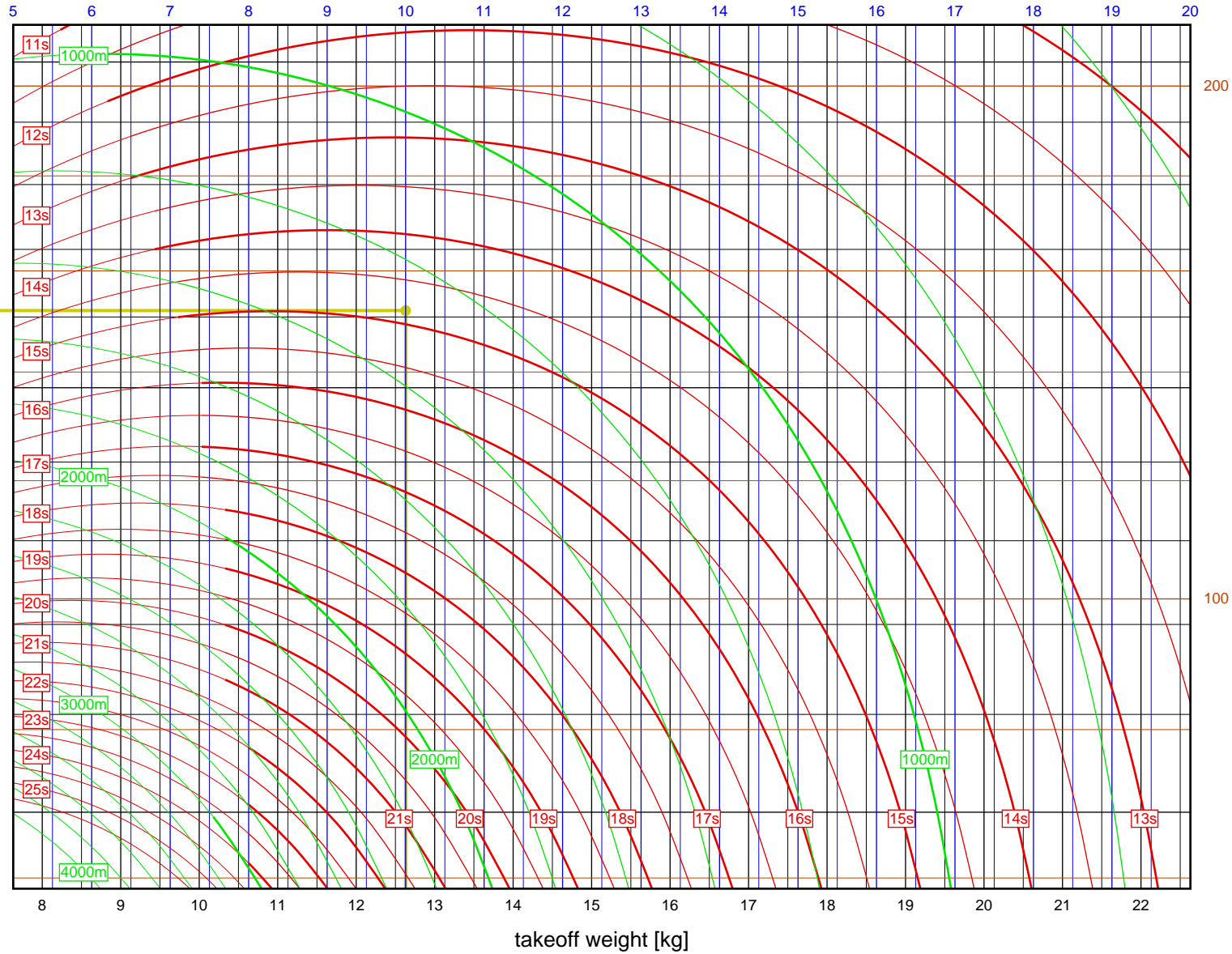
Data source:
Aerotech



1. From rocket diameter scale move down along slanted line to vertical line matching drag coefficient.
2. Move along horizontal to left border of density scale
3. Move up slanted line to vertical line matching density at launch site
4. From intersection point move horizontally to vertical line matching rocket mass
5. Read off expected time to apogee from red curves, altitude from green curves

Sample: diameter = 152mm, drag = 0.65, density = 1180 g/m³, weight = 12.630kg
Results: time to apogee: 14.9s, expected altitude: 1278m

empty weight [kg]



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L2300G

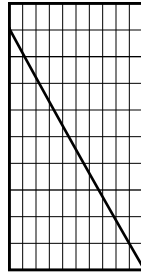
L2300G

6", K-L⁸

Kosdon-by-Aerotech M2900R

I_{tot} = 5460.9 Ns
 F_{avg} = 2264.2 N
 t_{burn} = 2.41 s
 d = 75 mm

Data source:
Aerotech



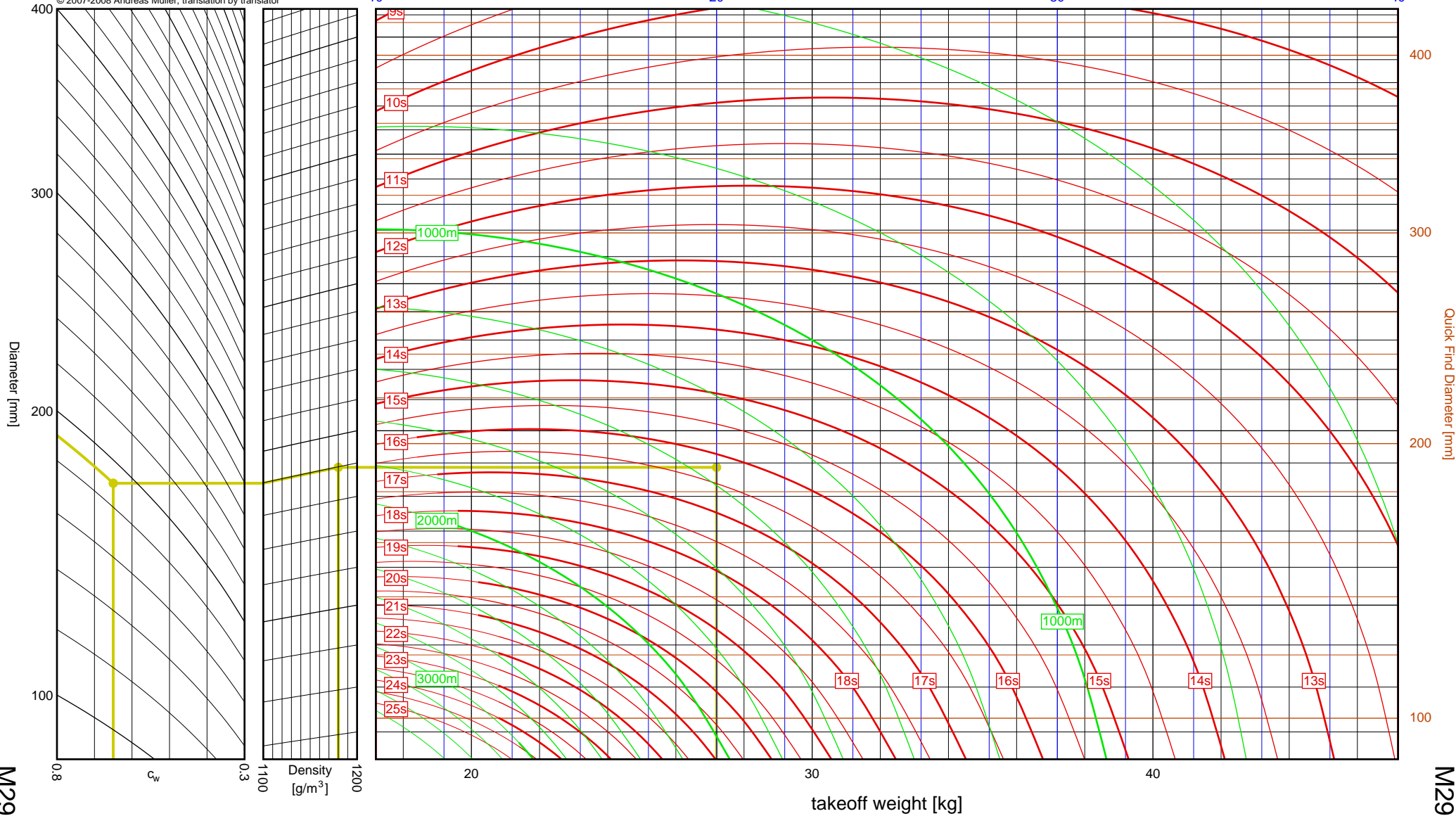
Launch site altitude [m ASL]

1. From rocket diameter scale move down along slanted line to vertical line matching drag coefficient.
2. Move along horizontal to left border of density scale
3. Move up slanted line to vertical line matching density at launch site
4. From intersection point move horizontally to vertical line matching rocket mass
5. Read off expected time to apogee from red curves, altitude from green curves

Sample: diameter = 190mm, drag = 0.65, density = 1180 g/m³, weight = 27.195kg
Results: time to apogee: 16.2s, expected altitude: 1359m

empty weight [kg]

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M2900R

M2900R

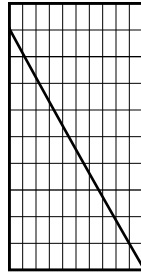
7.5" ^s

Kosdon-by-Aerotech M3500R

I_{tot} = 7312.0 Ns
 F_{avg} = 3111.6 N
 t_{burn} = 2.35 s
 d = 75 mm

Data source:
Aerotech

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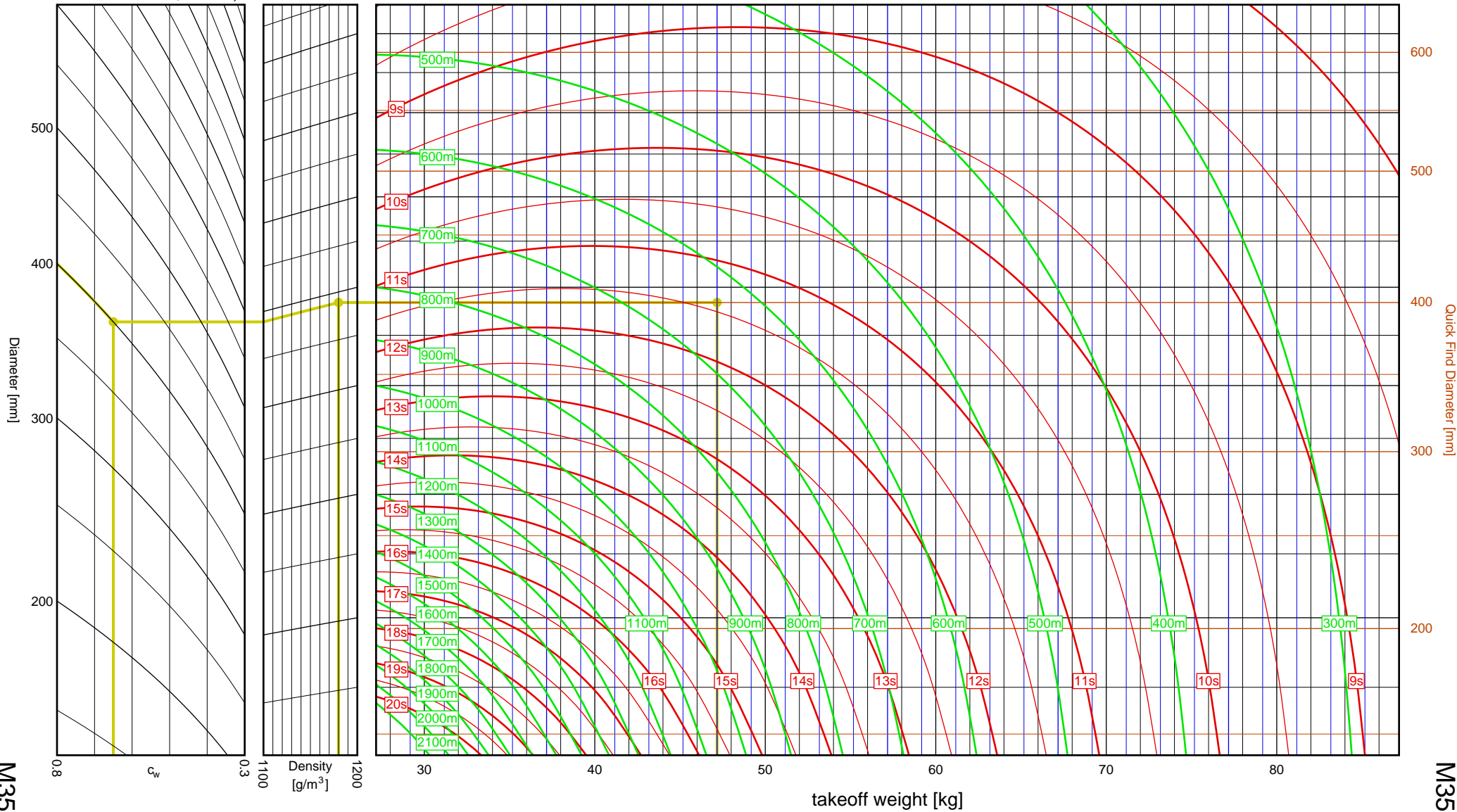


Launch site altitude [m ASL]

1. From rocket diameter scale move down along slanted line to vertical line matching drag coefficient.
2. Move along horizontal to left border of density scale
3. Move up slanted line to vertical line matching density at launch site
4. From intersection point move horizontally to vertical line matching rocket mass
5. Read off expected time to apogee from red curves, altitude from green curves

Sample: diameter = 400mm, drag = 0.65, density = 1180 g/m³, weight = 47.173kg
Results: time to apogee: 11.4s, expected altitude: 633m

empty weight [kg]



M3500R

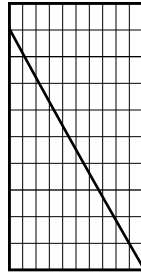
M3500R

Kosdon-by-Aerotech

M1450W

I_{tot} = 8069.7 Ns
 F_{avg} = 1354.0 N
 t_{burn} = 5.96 s
 d = 75 mm

Data source:
Aerotech

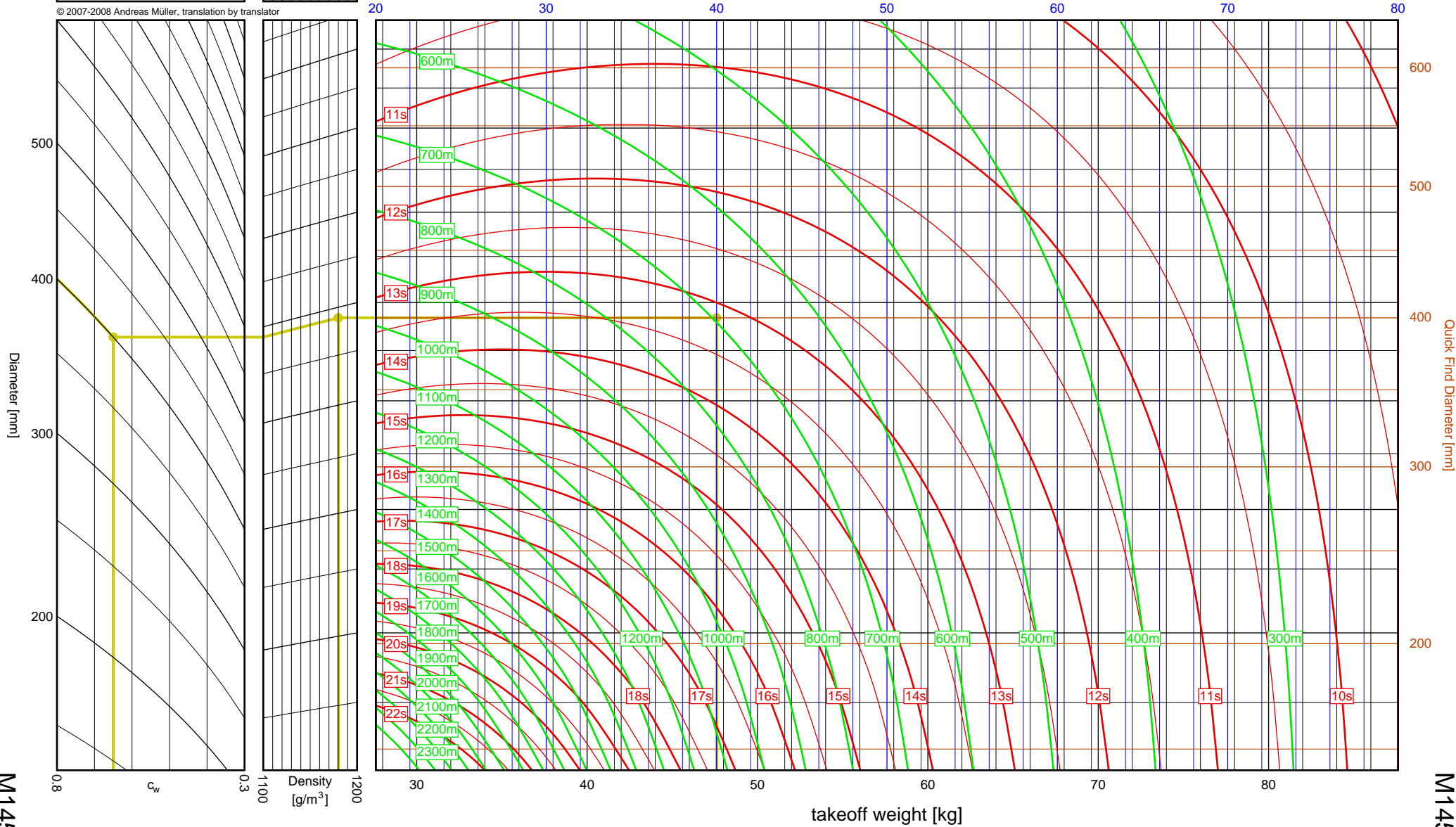


Launch site altitude [m ASL]

1. From rocket diameter scale move down along slanted line to vertical line matching drag coefficient.
2. Move along horizontal to left border of density scale
3. Move up slanted line to vertical line matching density at launch site
4. From intersection point move horizontally to vertical line matching rocket mass
5. Read off expected time to apogee from red curves, altitude from green curves

Sample: diameter = 400mm, drag = 0.65, density = 1180 g/m³, weight = 47.600kg
Results: time to apogee: 13.1s, expected altitude: 696m

empty weight [kg]



M1450W

M1450W

G135R	3-1
G82W	3-2
H130W	4-2
H225R	4-1
I170S	4-3, 5-1, 6-1
I301W	4-4, 5-2, 6-2
I550R	4-5, 5-3, 6-3
J740G	5-4, 6-4, 7-2
K1750R	8-2
K520F	7-1
K700F	7-3, 8-1
K750W	8-3
L2300G	8-4
M1450W	10-2
M2900R	9-1
M3500R	10-1