

Rocket Trajectory Nomograms

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Model rockets, A-C

Model rockets D-E

Mid Power, F-G

Small High Power, H-I

2inch High Power, I-J

3inch High Power, I-J

4inch High Power, J-K

6inch High Power, K-L

7.5inch High Power

Large rockets

A-C

D-E

F-G

H-I

2", I-J

3", I-J

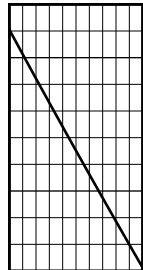
4", J-K

6", K-L

7.5"

X

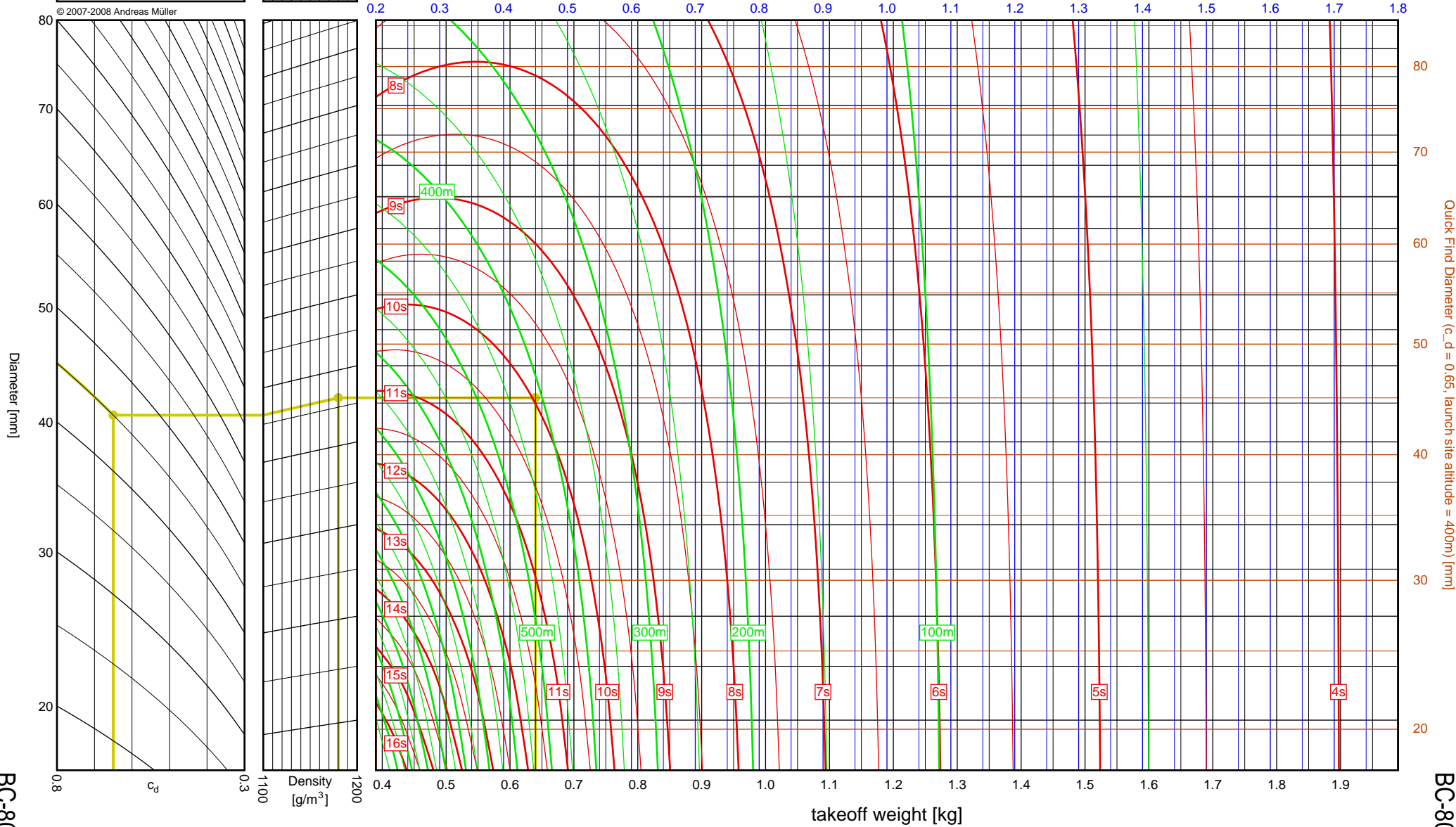
Maurer	
BC-80_SB	
I_{tot}	= 73.5 Ns
F_{avg}	= 28.2 N
t_{burn}	= 2.60 s
d	= 38 mm
Data source: http://www.raketenmodellbau.org	



1. From rocket diameter scale move down along slanted line to vertical line matching drag coefficient.
2. Move along horizontal to left border of density scale
3. Move up slanted line to vertical line matching density at launch site
4. From intersection point move horizontally to vertical line matching rocket mass
5. Read off expected time to apogee from red curves, altitude from green curves

Example: diameter = 45mm, drag = 0.65, density = 1180 g/m³, weight = 0.640kg
Results: time to apogee: 10.0s, expected altitude: 410m

empty weight [kg]



3-1

Quick Find Diameter ($c_d = 0.65$, launch site altitude = 400m) [mm]

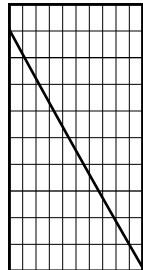
BC-80_SB

BC-80_SB

Maurer BC-125"4

$I_{tot} = 174.4 \text{ Ns}$
 $F_{avg} = 40.2 \text{ N}$
 $t_{burn} = 4.34 \text{ s}$
 $d = 38 \text{ mm}$

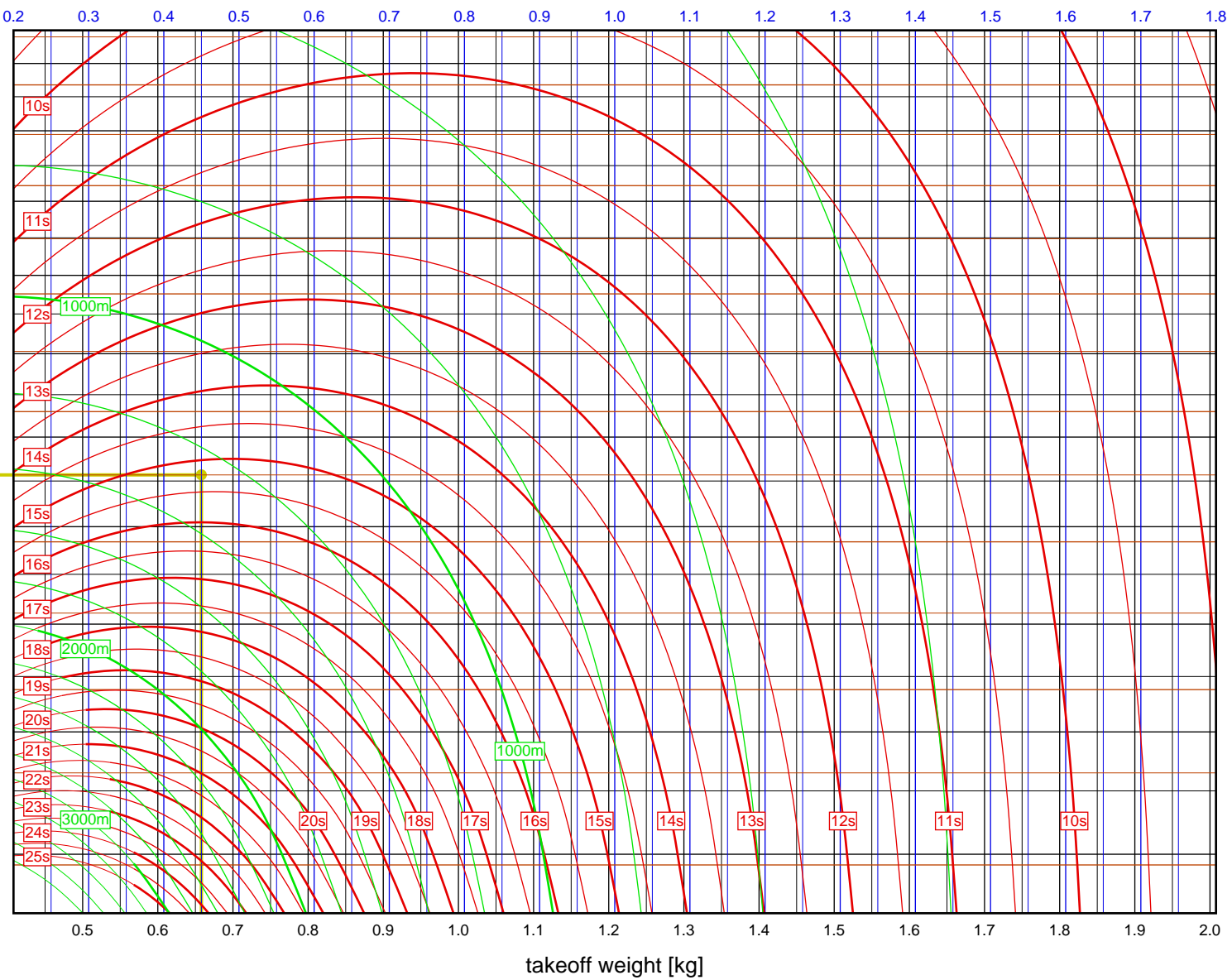
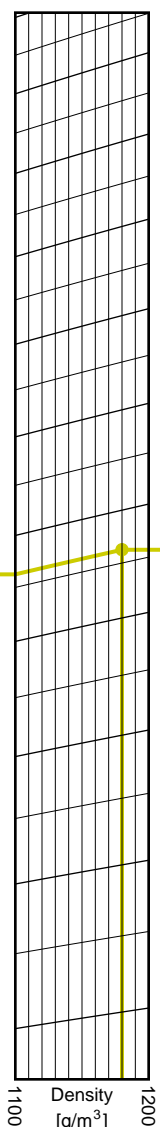
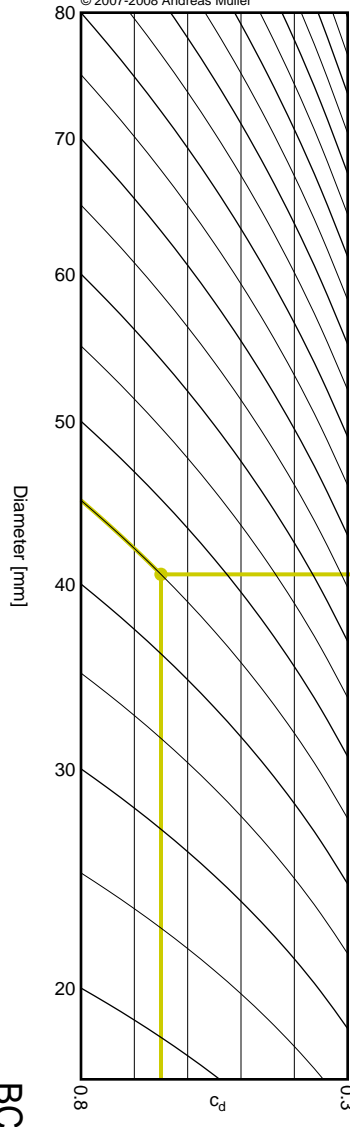
Data source:
<http://www.raketenmodellbau.org>



Launch site altitude [m ASL]

1. From rocket diameter scale move down along slanted line to vertical line matching drag coefficient.
 2. Move along horizontal to left border of density scale
 3. Move up slanted line to vertical line matching density at launch site
 4. From intersection point move horizontally to vertical line matching rocket mass
 5. Read off expected time to apogee from red curves, altitude from green curves
- Example:** diameter = 45mm, drag = 0.65, density = 1180 g/m³, weight = 0.658kg
Results: time to apogee: 15.2s, expected altitude: 1261m

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empty weight [kg]

takeoff weight [kg]

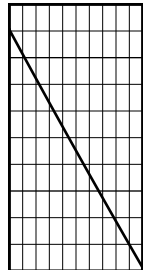
BC-125"4

BC-125"4

Maurer BC-125"6

$I_{tot} = 177.9 \text{ Ns}$
 $F_{avg} = 29.8 \text{ N}$
 $t_{burn} = 5.97 \text{ s}$
 $d = 38 \text{ mm}$

Data source:
<http://www.raketenmodellbau.org>

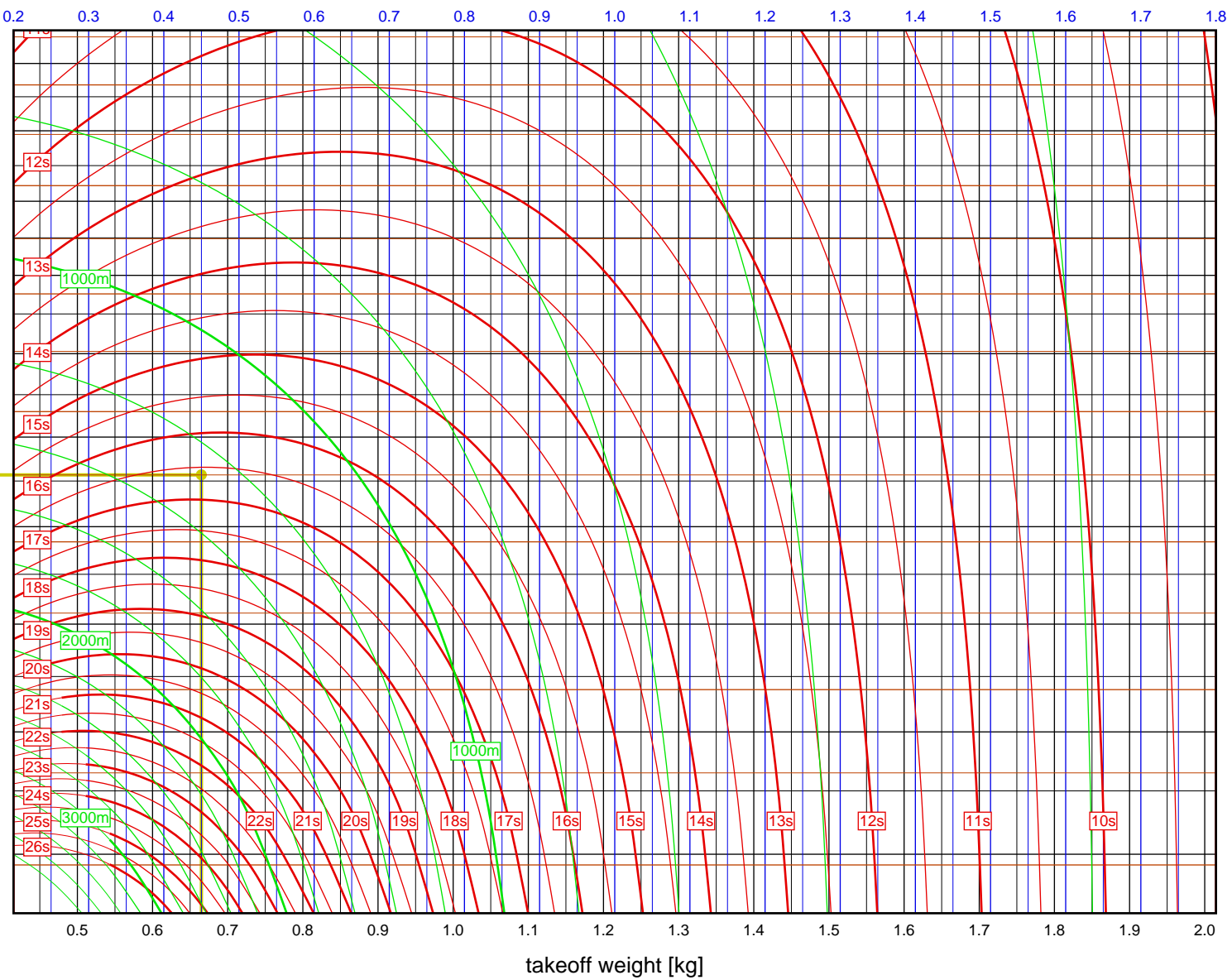
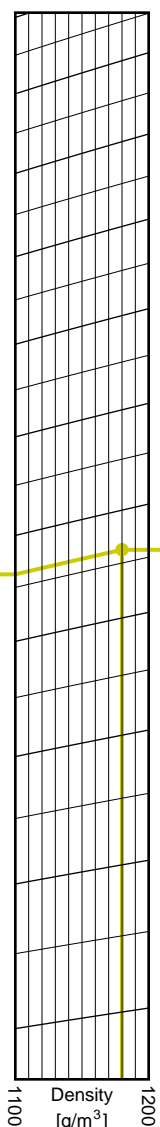
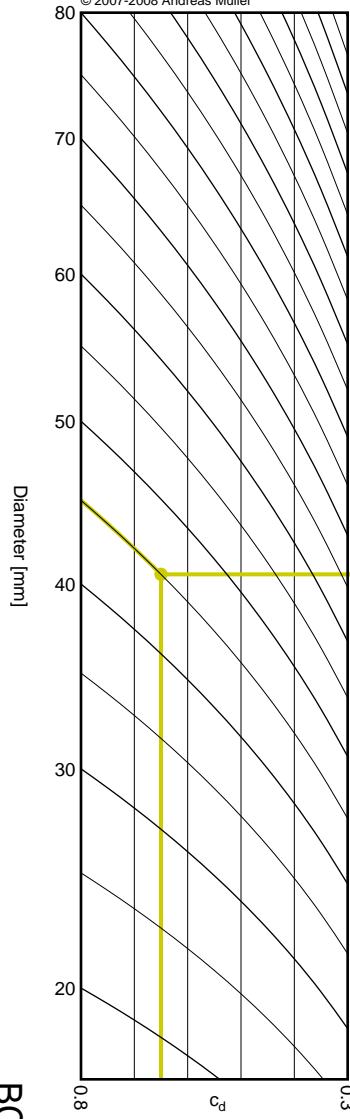


Launch site altitude [m ASL]

1. From rocket diameter scale move down along slanted line to vertical line matching drag coefficient.
2. Move along horizontal to left border of density scale
3. Move up slanted line to vertical line matching density at launch site
4. From intersection point move horizontally to vertical line matching rocket mass
5. Read off expected time to apogee from red curves, altitude from green curves

Example: diameter = 45mm, drag = 0.65, density = 1180 g/m³, weight = 0.665kg
 Results: time to apogee: 16.6s, expected altitude: 1269m

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empty weight [kg]

takeoff weight [kg]

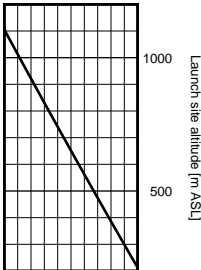
BC-125"6

BC-125"6

Maurer BC-125"1

$I_{tot} = 157.8 \text{ Ns}$
 $F_{avg} = 143.3 \text{ N}$
 $t_{burn} = 1.10 \text{ s}$
 $d = 38 \text{ mm}$

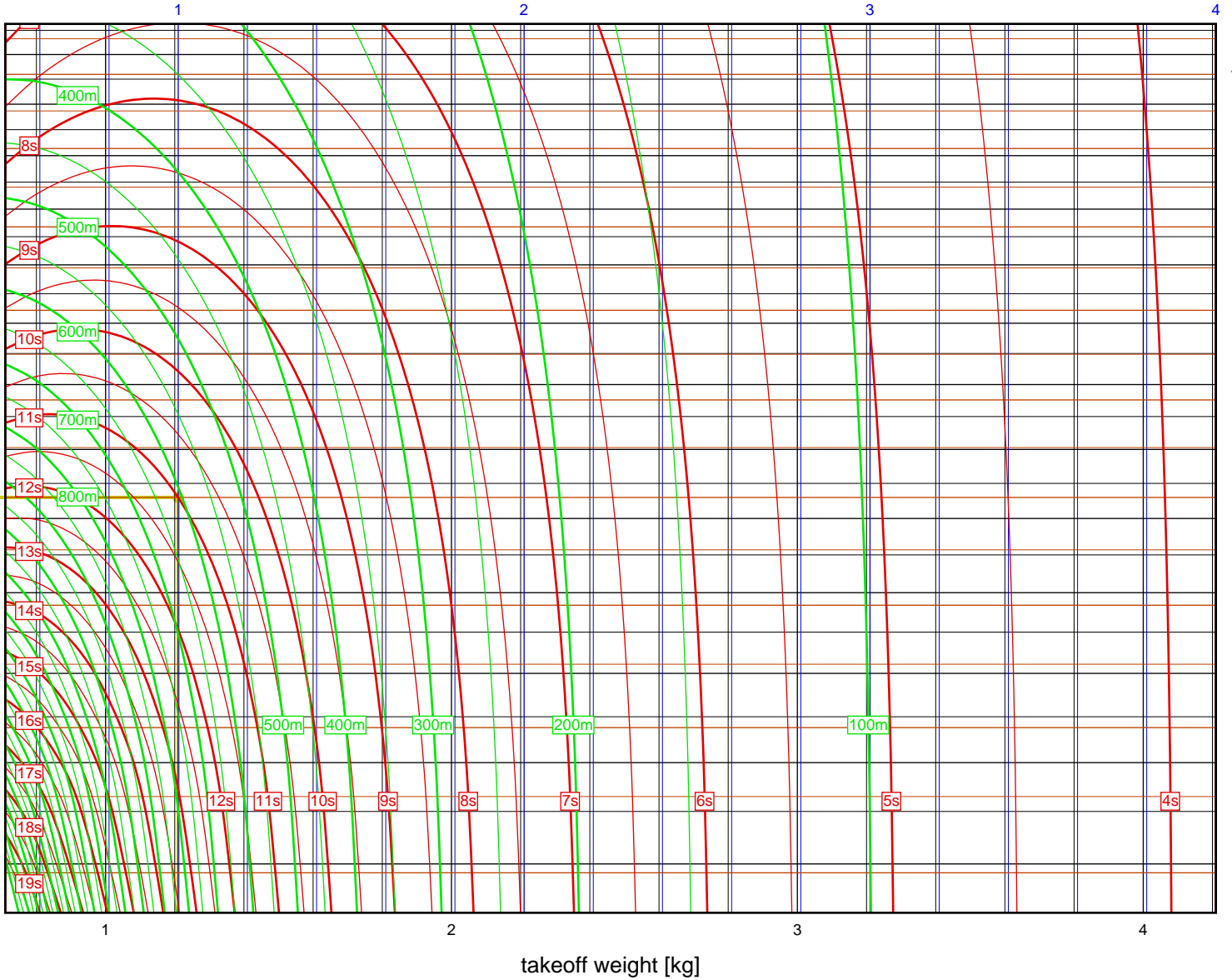
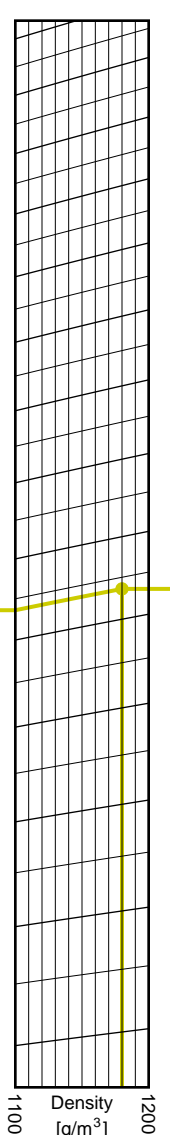
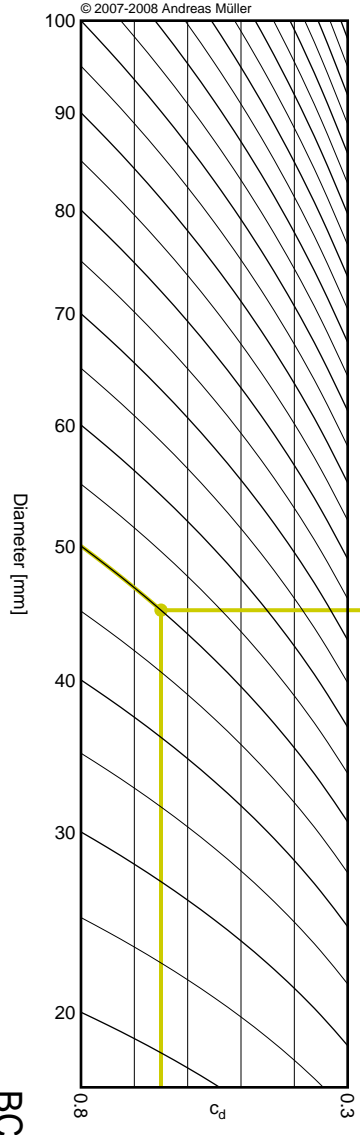
Data source:
<http://www.raketenmodellbau.org>



1. From rocket diameter scale move down along slanted line to vertical line matching drag coefficient.
2. Move along horizontal to left border of density scale
3. Move up slanted line to vertical line matching density at launch site
4. From intersection point move horizontally to vertical line matching rocket mass
5. Read off expected time to apogee from red curves, altitude from green curves

Example: diameter = 50mm, drag = 0.65, density = 1180 g/m³, weight = 1.210kg
 Results: time to apogee: 11.0s, expected altitude: 607m

empty weight [kg]

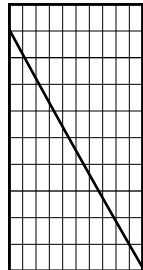


Quick Find Diameter ($c_d = 0.65$, launch site altitude = 400m) [mm]

BC-125"1

BC-125"1

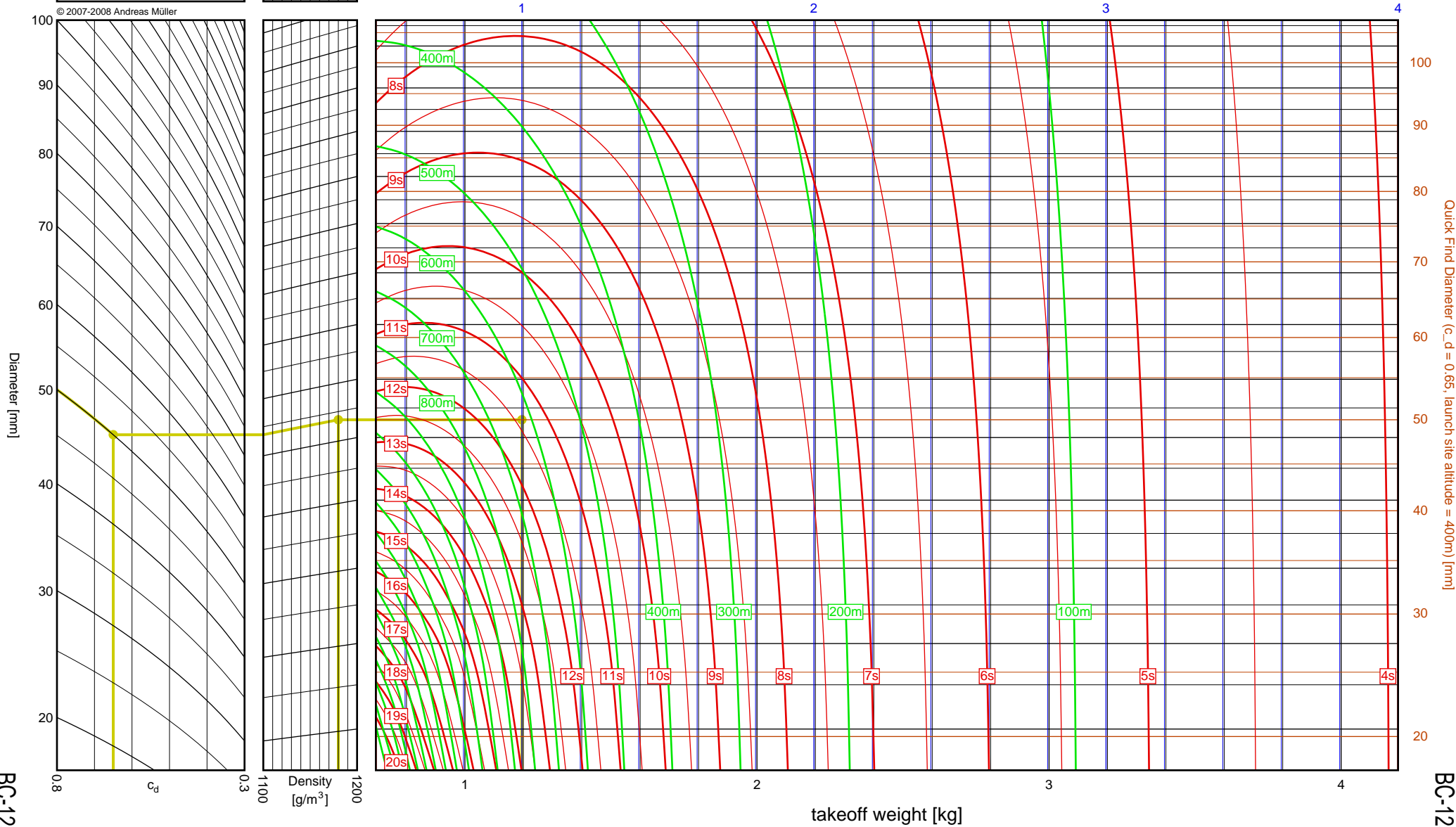
Maurer	
BC-125_Alu	
I_{tot}	= 161.2 Ns
F_{avg}	= 95.8 N
t_{burn}	= 1.68 s
d	= 38 mm
Data source: http://www.raketenmodellbau.org	



1. From rocket diameter scale move down along slanted line to vertical line matching drag coefficient.
2. Move along horizontal to left border of density scale
3. Move up slanted line to vertical line matching density at launch site
4. From intersection point move horizontally to vertical line matching rocket mass
5. Read off expected time to apogee from red curves, altitude from green curves

Example: diameter = 50mm, drag = 0.65, density = 1180 g/m³, weight = 1.195kg
 Results: time to apogee: 11.4s, expected altitude: 623m

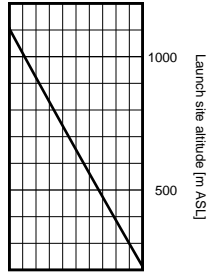
empty weight [kg]



BC-125_Alu

BC-125_Alu

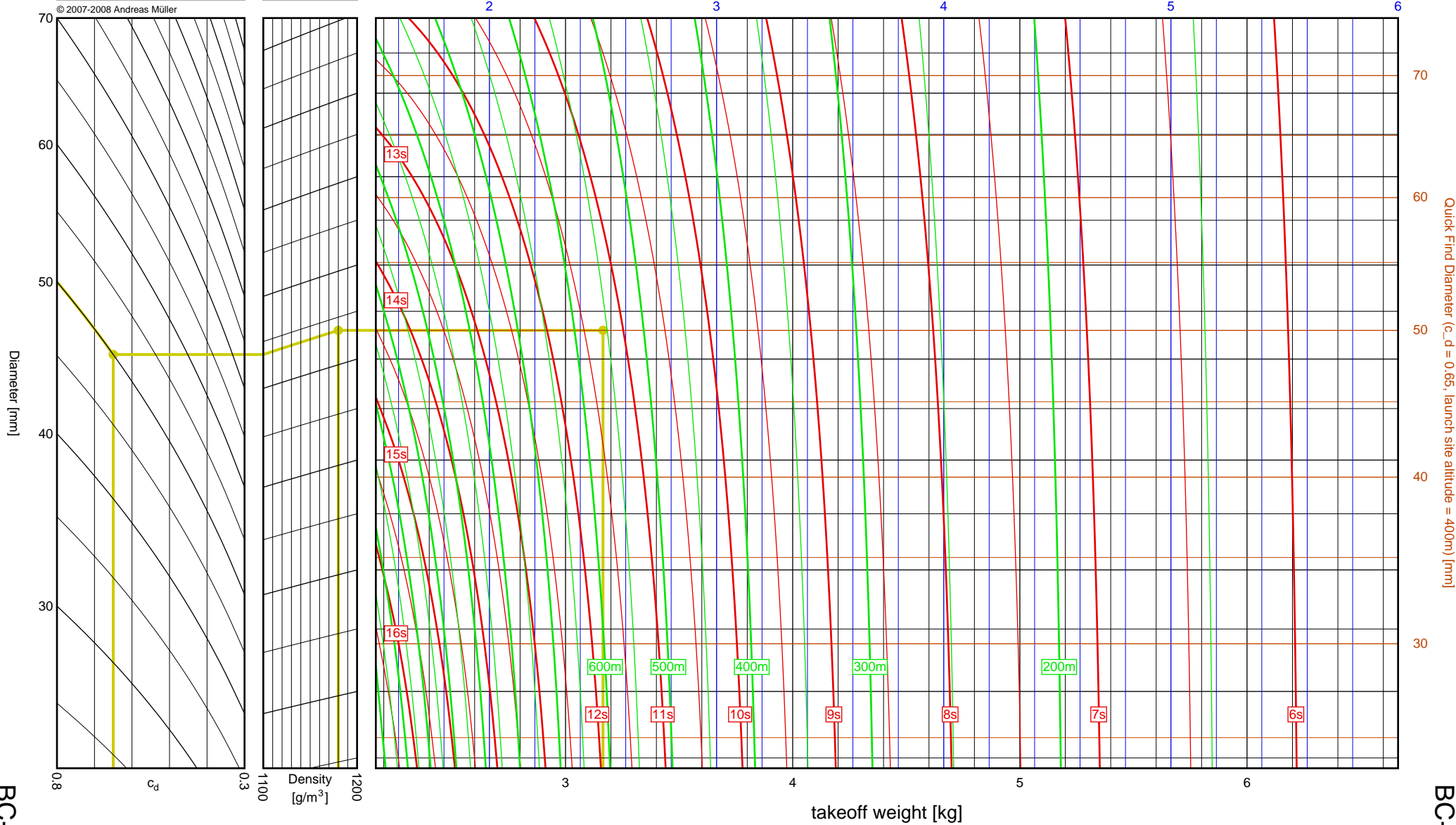
Maurer BC-360	
I_{tot}	= 356.5 Ns
F_{avg}	= 210.6 N
t_{burn}	= 1.69 s
d	= 72 mm
Data source: http://www.raketenmodellbau.org	



1. From rocket diameter scale move down along slanted line to vertical line matching drag coefficient.
2. Move along horizontal to left border of density scale
3. Move up slanted line to vertical line matching density at launch site
4. From intersection point move horizontally to vertical line matching rocket mass
5. Read off expected time to apogee from red curves, altitude from green curves

Example: diameter = 50mm, drag = 0.65, density = 1180 g/m³, weight = 3.165kg
 Results: time to apogee: 11.3s, expected altitude: 556m

empty weight [kg]



BC-360

BC-360

2", I-J

BC-125"1	4-1
BC-125"4	3-2
BC-125"6	3-3
BC-125_Alu	4-2
BC-360	5-1
BC-80_SB	3-1